A horizontal cryogenic superconducting inertial sensor J.V. van Heijningen^{1,†}, F. Badaracco¹, E.C. Ferreira¹, A. Perali²

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Abstract

We present an inertial sensor that will exploit our efforts towards cryogenic test masses in gravitational wave (GW) detectors. The superconducting metals or high-T_c materials give access to low-loss actuators, mechanics and readouts. This inertial sensor can be used in the Einstein Telescope (ET) to monitor the effects on cryo-cooling the final stages or can be deployed on the Moon to directly detect GWs. Around 1 Hz, modelled sensitivity is about 3 orders of magnitude better than state-of-the-art if the sensor is cooled below 45 K (e.g. 50% YBCO T_c).



Main features of the sensor

A cryogenic superconducting inertial sensor (CSIS) [1] is shown in fig. 1. Expected sensitivity curves for different mechanics material operating in cryogenic conditions are shown in fig. 2. The design features:

- \blacktriangleright pm /VHz readout to take out proof mass mode;
- > superconducting thin film spiral coils for low voltage and low heat load/power consumption;
- Iow power consuming custom CMOS cryo-chip for readout, coil driving controls;



Figure 2: Minimal detectable inertial displacement comparison between two CSIS options (see figure 5 for more detail) at 5 K and the measured state-of-the-art used in gravitational wave detectors today. The lowest geo-seismicity are summarized by the Peterson low noise model (LNM) [3].



Figure 1: CSIS features a Watt's linkage with 2 superconducting actuators, large range (as large as the mask) Rasnik [2] readout and high-frequency precision interferometric readout. Green arrows indicate how connected parts move. The piezo can be used for calibration or readout.

- \succ fm /VHz interferometric readout;
- polarizing optics directing all light to photodiodes for low heat load;
- \succ tuning mass to set Watt's linkage f₀.

... in Einstein Telescope

A comparison of the low frequency ET (ET-LF) and LIGO is shown is fig. 3. CSIS, with in fm/ \sqrt{Hz} sensitivity below 1 Hz, can monitor the cold platform from which the mirror is suspended and project down. Depending on the ultimate suspension design it may contribute to the control. The cold platform may be at 5 K, which opens up the superconductive regime for niobium. If the mirror temperatures are higher, then high-T_c materials are used and coils in actuators surfaces.

Figure 3: Mirror displacement sensitivity for a) Advanced LIGO during the first detections, adapted from [4]. Panel b) shows the ET-LF cryogenic low-frequency design, adapted from [5]. A pink dotted line shows alignment with panel a). Note a factor 10⁴ improvement is needed between currently obtained and future modelled displacement sensitivity.

pick-up coils on both proof mass sides or superconducting reentrant [11] cavities (famous for their use in bar detectors [12]) hold a promise for subfm/VHz sensitivity.

Conclusion

CSIS, when deployed on the Moon and in ET, may enable GW science between 1 mHz and 5 Hz. Within the E-TEST suspension prototype, development is ongoing.

References

[1] J.V. van Heijningen, <u>JINST 15 P06034</u> (2020) [2] H. van der Graaf et. al., <u>arXiv:2104.03601</u> (2021) [3] J. Peterson, <u>US Geol. Surv. 1-95</u>, pp. 93-322 (1993) [4] B. P. Abbott et al. , <u>Phys. Rev. Lett. 116, 131103</u> (2016) [5] S. Hild et al., <u>Class. Quant. Grav. 27 015003</u> (2010) [6] Jan Harms et al., <u>Astrophysical Journal 9101</u> (2021)

...on the Moon

Using the Moon as a bar detector, the Lunar Gravitational-wave Antenna [6] (LGWA) will bridge the 0.12-2 Hz gap between the Laser Interferometric Space Antenna (LISA) and ET. CSIS will detect lunar surface motion caused by the passing GW. In the case of a niobium CSIS, low-vibration cryocoolers (e.g. sorption coolers [7]) are Using high-T_c needed. materials allows use of permanent shadows of lunar South pole craters as natural cryostats. LGWA details on appendix



Figure 5: Seismically isolated table using pendulums and Euler springs [8] with a cryogenics for sensor characterisation. Via helium conduction [9] thermal flex links and suspended rods/ cylinders [10], the cold heads are decoupled from thermal shields. Tilt subtraction not shown.

poster. To reach the sensitivities as shown in fig. 4, we need a huddle test on a seismically isolated platform with a low-vibration cryogenics as shown in fig. 5.

Future improvements

High-Q silicon Watt's linkages can't be



Figure 4: Noise budget CSIS for silicon as mechanics material ($Q = 10^6$ and T = 5 K); niobium thermal noise (Q = 10⁴) is also plotted. A 5 mN/A actuator is modelled as driven by a 10 μ V/Hz DAC. We model frequency noise as $500/\sqrt{f}$ using a conservative static arm length difference of 0.5 mm. The injected power is 25 mW, the proof mass is 1 kg and f_0 is tuned to 0.1 Hz.

monolithic because of flexure surface losses when machined. An assembly is therefore strategy (see fig. 6) devised. Other readout strategies than interferometric ones are investigated as well. SQUID readouts using current





Figure 6: Assembly procedure for a silicon CSIS. A frame (in 2 pieces) and proof mass is cut out of silicon block using EDM. The leg and flexures are etched out of a thick 500 μ m high-quality wafer. Parts are hydro catalysis bonded (HCB) to form a low mechanical loss Watt's linkage.





Figure 1: The simplified GW response model shown in red used here. Normal-mode Figure 2: Average surface temperatures at the lun sum truncated at n = 22 and models only include spheroidal quadrupole modes. showing possible deployment in the 40 K perman



lander/ central station

1 of 4 seismic stations

kilometre scale array in arbitrary size crater

Figure 3: Configuration of the kilometre scale LGWA array. Lander position is indicated for the case when all seismic stations are deployed from a single landing site; separate landings are possible. Solar power is transmitted to array via laser.

> Maximally soft lunar model 10⁴



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10 ⁻¹⁵		
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	Figure 4: Predicted LGWA strain spectral density con and Einstein Telescope (ET). Cyan highlights where L	n G
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npared to the Laser Interferometer Space Antenna (LISA) GWA bridges the 0.12 – 2 Hz gap between LISA and ET.

vitational waves on the Moon

astic body excited by gravitational waves was one on of gravitational waves (GWs). The Earth-bound W frequencies due to their dimensions. Shown in z frequencies and placing seismometers on the smometers require stable temperatures, obtained ater, a mission sketch is shown in fig. 3. A central distributes it to the seismic stations. These four sing local seismic signals from meteor impacts. WA seismometer sensitivity divided by the Moon and ET. In addition, the extreme sensitivity of the physics) using said meteors as active sources.