

Helium-based cooling concept of the ET-LF interferometer

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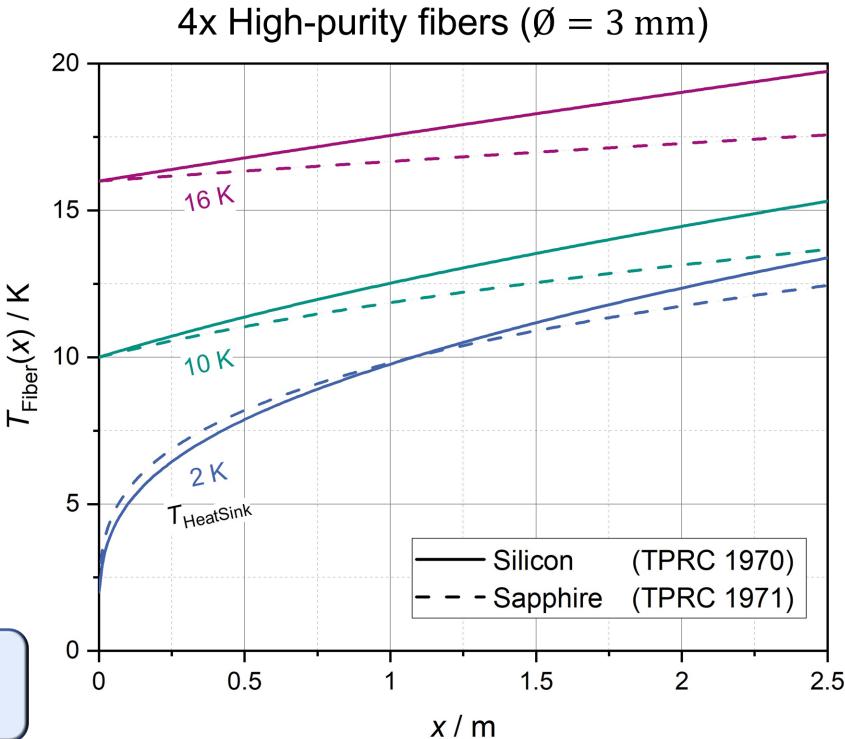
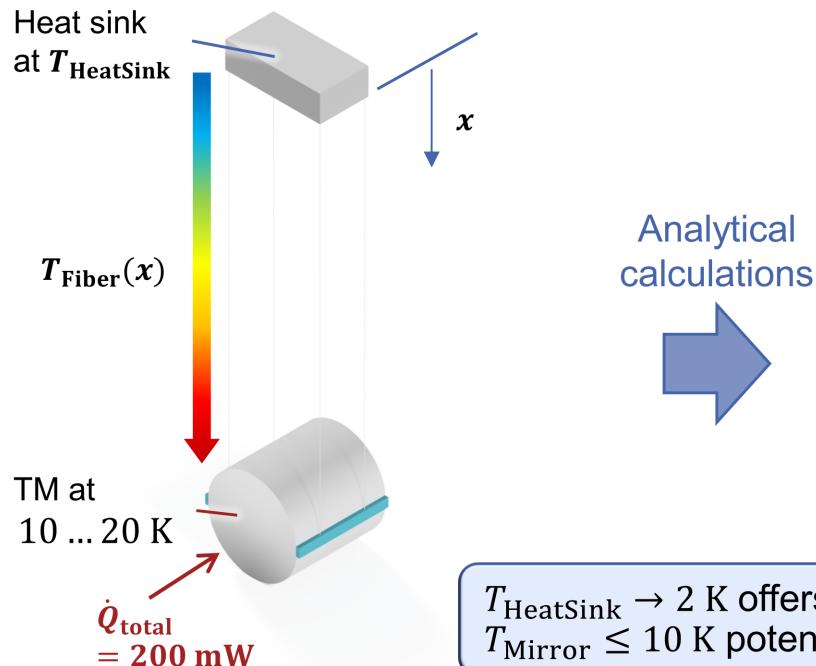
Gravitational Wave
Advanced Detector Workshop
17-21 May 2021



Image: @EGO PHOTO / CC BY (cropped)

Test mass temperature limitation

Last-stage suspension scheme:



He-II: payload heat extraction

Two liquid phases of ^4He :

- He-I (classical liquid helium)

 - Behaviour: ~ideal gas

$$T_\lambda(1 \text{ atm}) \approx 2.17 \text{ K}$$

$$\begin{array}{l} T > T_\lambda \\ T < T_\lambda \end{array}$$

- He-II ("two fluid model" [1][2])

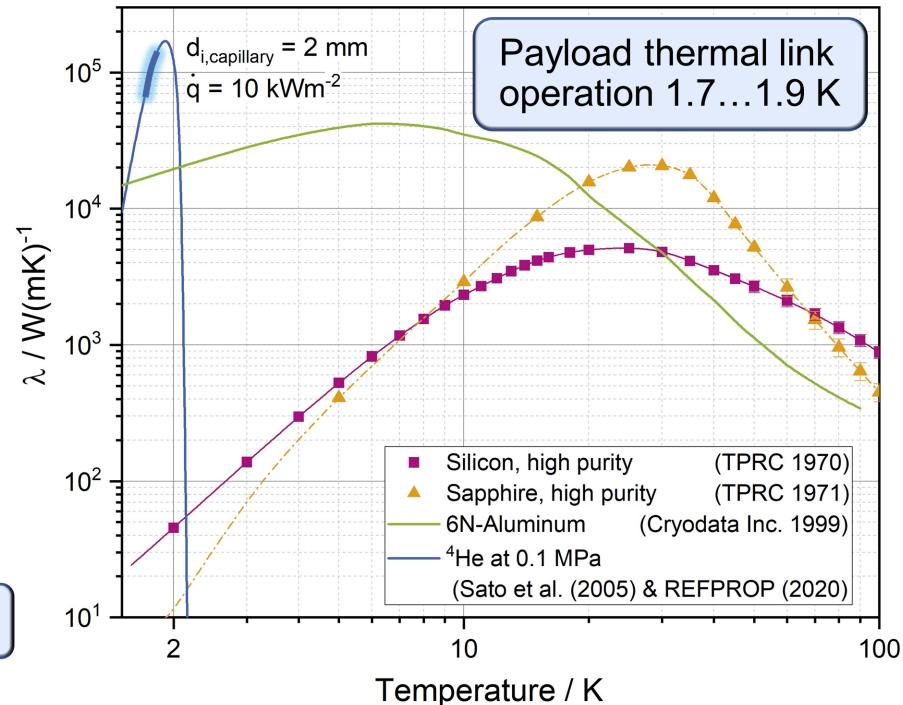
 - Normal component

 - Superfluid component

 - Bose-Einstein condensate

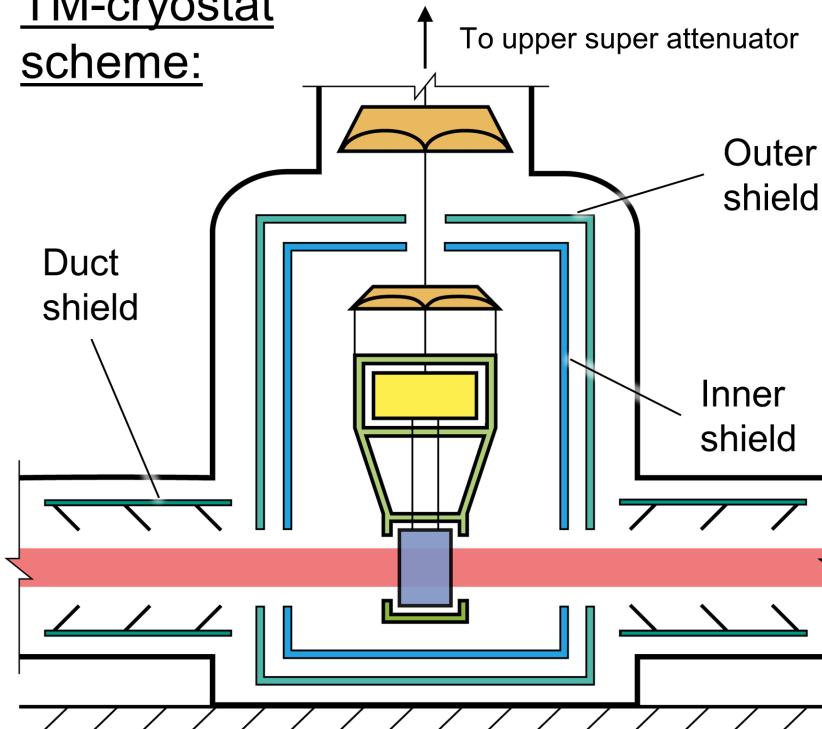
He-II: enhanced heat transfer properties

Sources: [1] Tisza, L. Transport Phenomena in Helium II. *Nature* 141, 913 (1938).
 [2] Landau, L. Theory of the Superfluidity of Helium II. *Phys. Rev.* 60, 356-358 (1941).



TM cryostat cooling: temperature levels

TM-cryostat scheme:



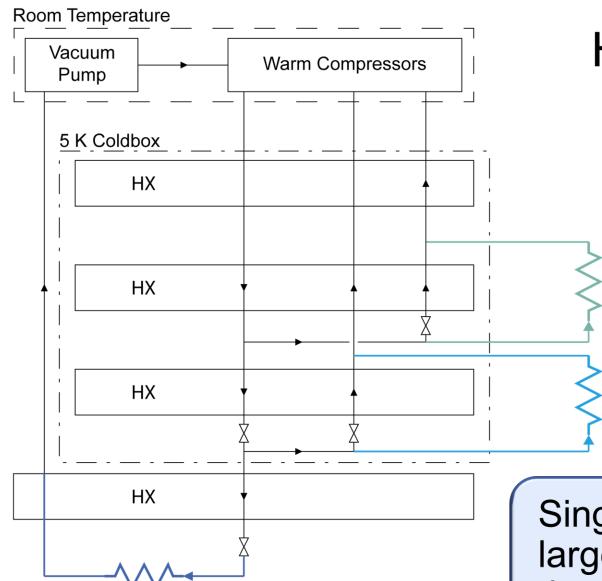
Three temperature stages:

Part(s)	Temp. level	Estimated cooling power
Outer thermal shield	50...80 K	x...10 ³ W
Inner thermal shield	5 K	x...10 ² W
Payload heat sink	2 K	x...1 W

$T_{\text{InnerShield}} < T_{\text{Mirror}} (\sim 10 \text{ K})$ possible

Helium-based cooling power provision

Basis: He-refrigerator + subcooler



He-supplies:

- 50...80 K (HP)
- 5 K (LP)
- 2 K (VLP)

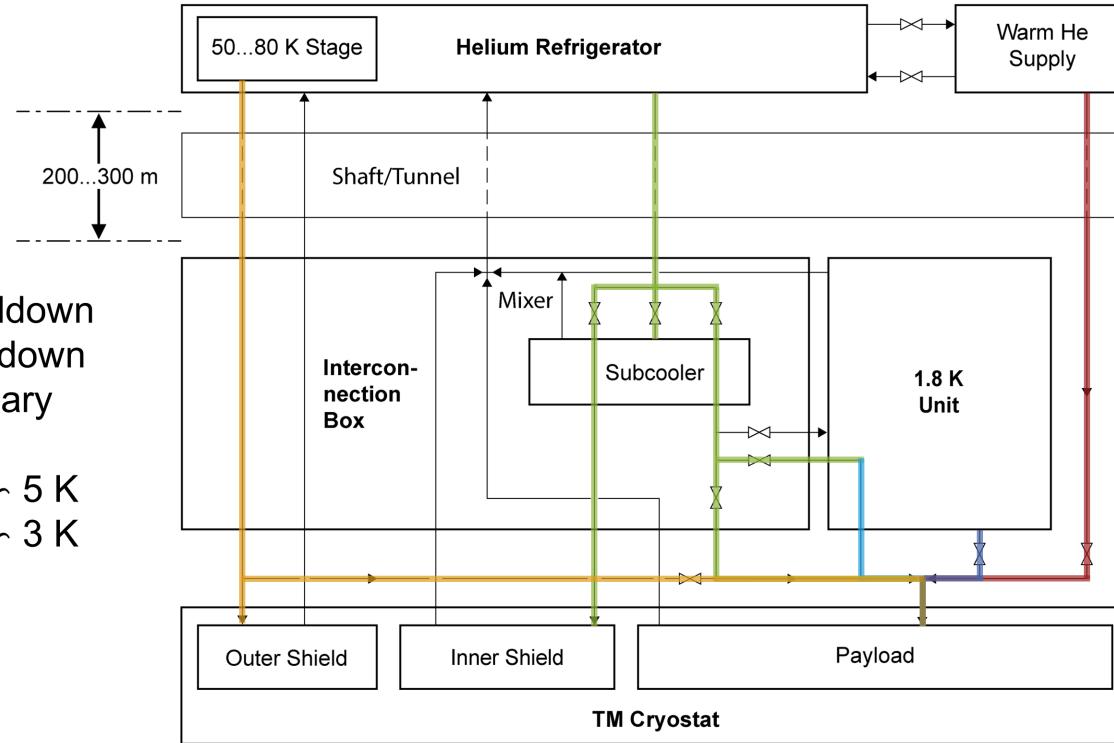
Example: Linde L-Series



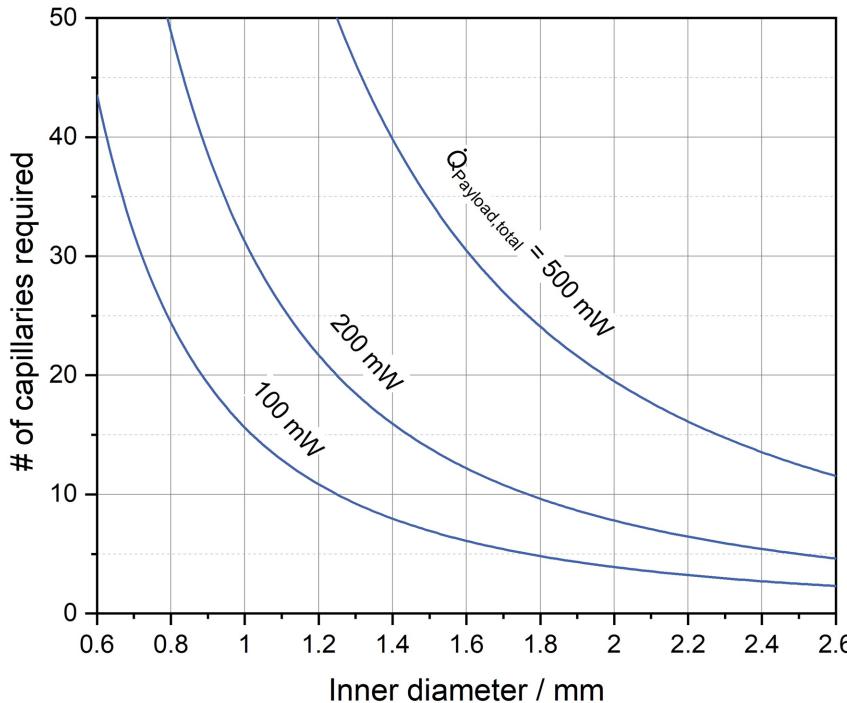
Image: L-Series - Standard Helium Liquefiers / Refrigerators, Linde Kryotechnik AG, 2021.

Basic layout: complete cryostat cooling

1. Outer shield cooldown
2. Inner shield cooldown
3. Payload preliminary cooldown
4. Payload link $\rightarrow \sim 5\text{ K}$
5. Payload link $\rightarrow \sim 3\text{ K}$
6. He-II formation
7. Steady-state (no flow)



Payload thermal link: He-II capillaries



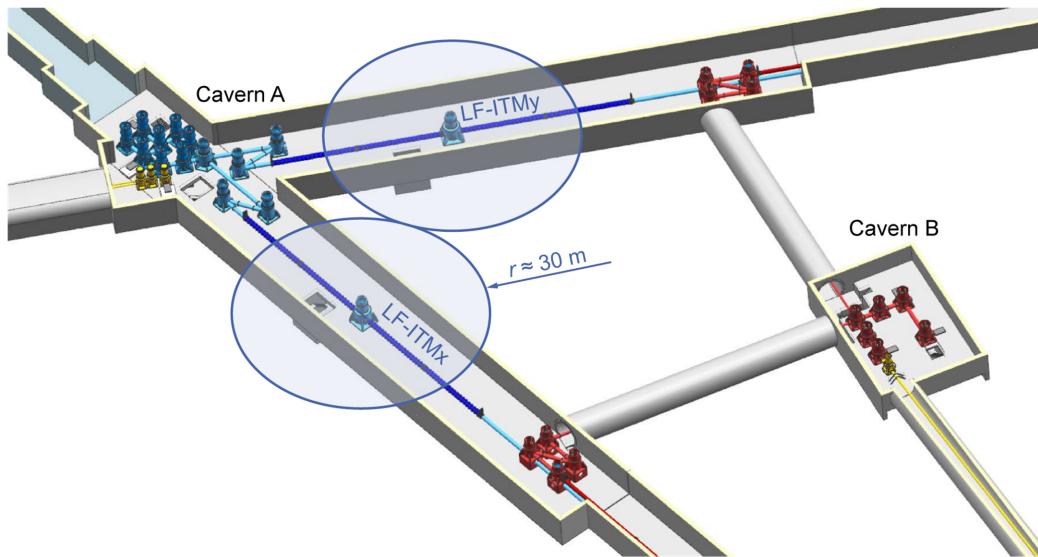
Key boundary conditions:

- Operating pressure: 0.1 MPa
- Capillary length: 30 m
- Capillary cold end temp. (He-II): 1.80 K
- Capillary warm end temp. (He-II): 1.90 K

Dimensioning example (approx.):
 10 capillaries with $d_i = 1.8 \text{ mm}$ can extract
 200 mW from a payload at 1.9 K over 30 m
 distance with a ΔT of only 0.1 K.

1.8 K unit positioning possibilities

Corner cavern scheme:



1.8 K Unit main components:

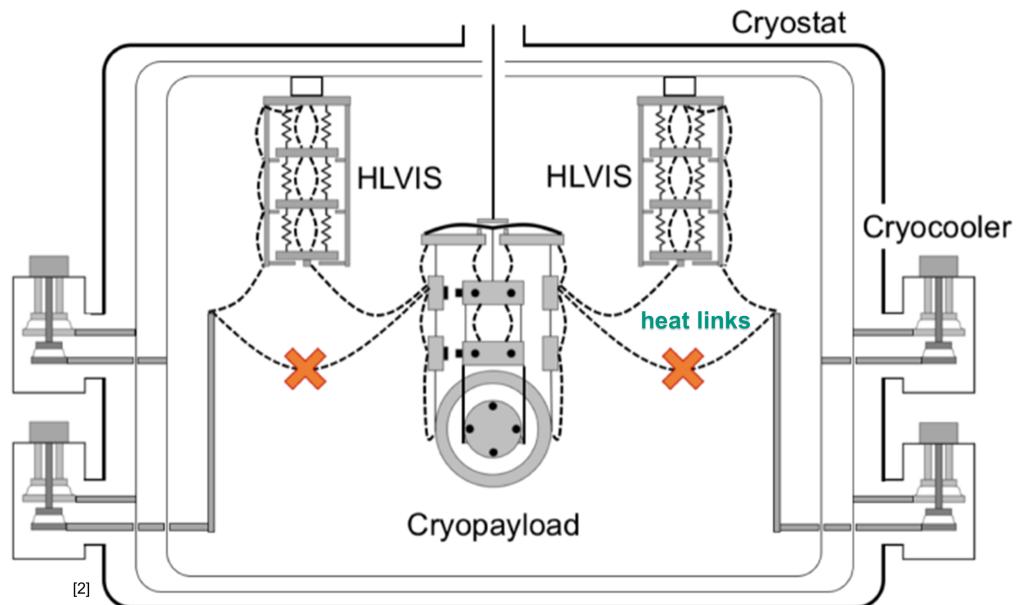
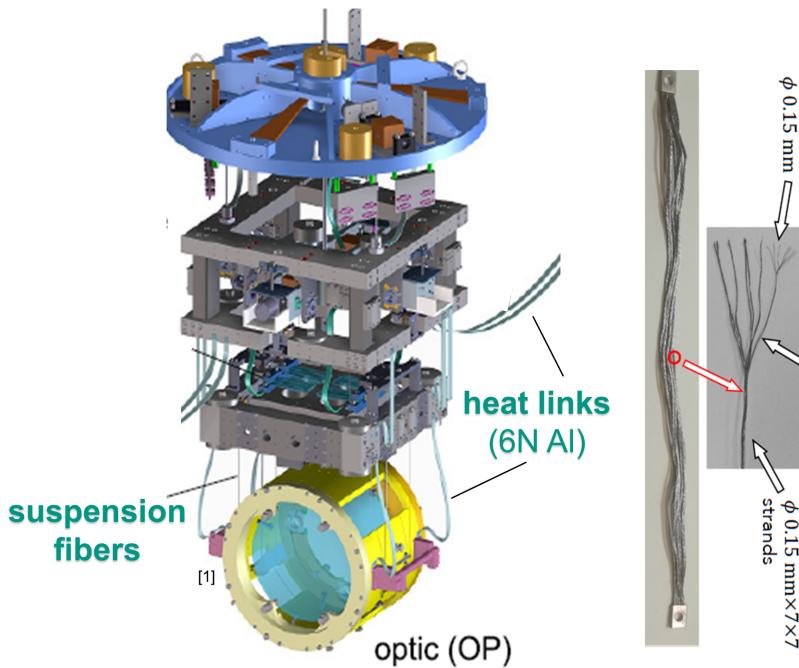
- Vacuum vessel
- Low-pressure heat exchanger
- Vacuum pumps

Long capillaries offer low-noise cooling potential and flexible positioning of the 1.8 K units.

Image: ET Steering Committee Editorial Team. Design Report Update for the Einstein Telescope. Technical report, ET-0007B-20, 2020 (altered)

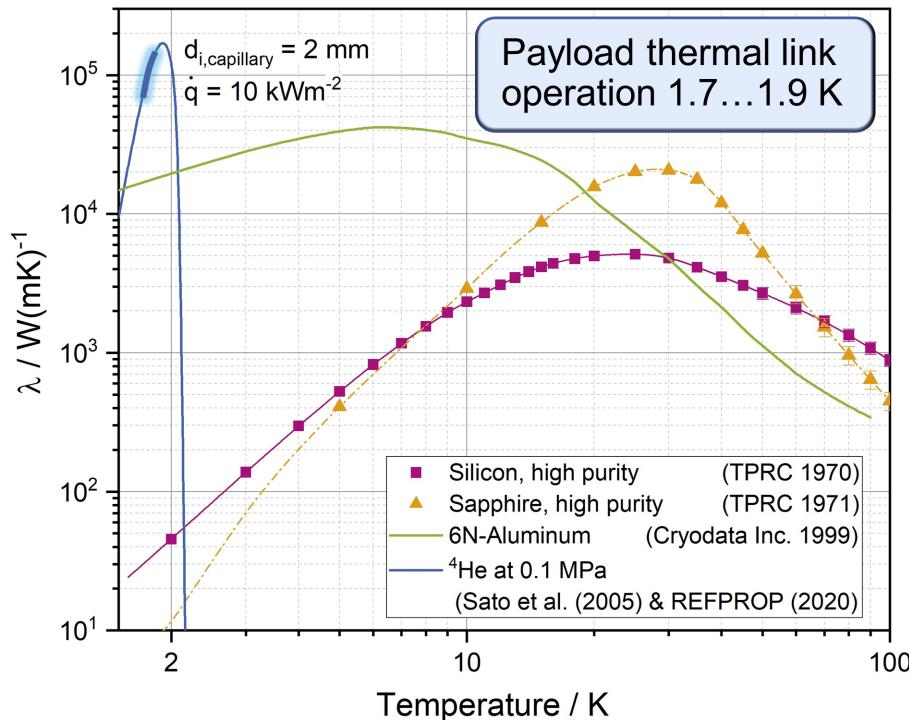
Vibration propagation via He-II hollow heat links

KAGRA experience on vibration transmission into payload



Sources: [1] Akutsu et al (2019) – First cryogenic test operation of underground km-scale GW Observatory KAGRA
[2] T.Yamada (2020) – Low-Vibration Conductive Cooling of KAGRA Cryogenic Mirror Suspension

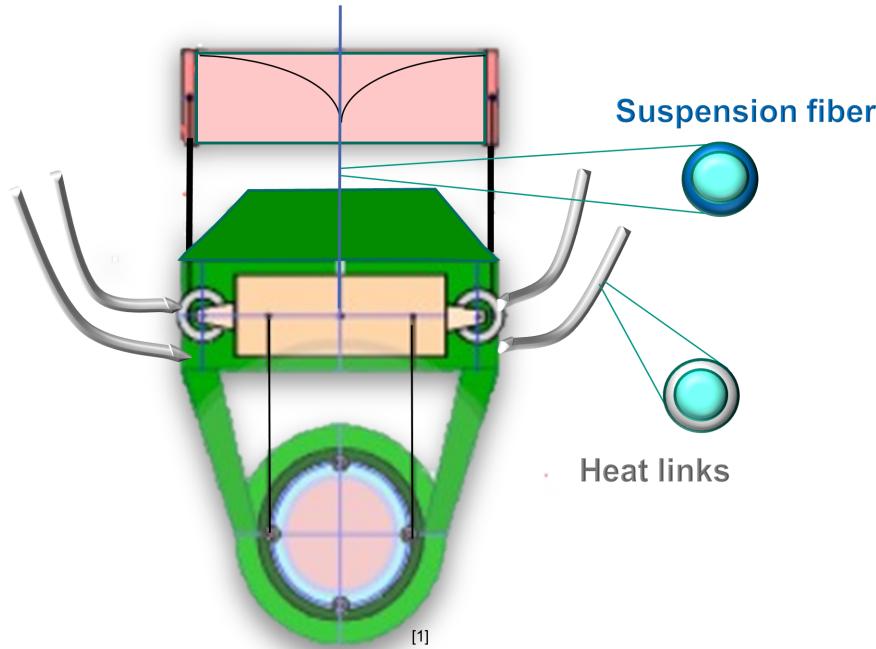
He-II hollow capillaries as heatlinks of ET-LF payload



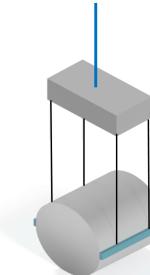
Lower thermal dissipations

- Cryogenic temperatures
- Dissipation-free superfluid component in He-II

Integration of He-II filled capillary into payload



- He-II filled capillary ($\varnothing \leq 3 \text{ mm}$)
- Hollow capillary as **suspension fiber**
 - Marionette
 - Mirror (?)
- Hollow capillary as external heat links



Source: [1] Payload design from P.Rapagnani: ET LF-Main Features and Constrains (26.04.21)

Theoretical description of thermal dissipation in He-II capillaries

Vibrational Noise into payload

- He-II capillaries → suspension fibers and/or heat links
- Cooling system noise

Experimental Proof of Concept

- Ultra-low noise He-cooling system

Thank you for your attention!

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