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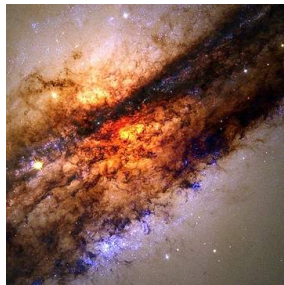


ULTRA-HIGH ENERGY COSMIC PARTICLES ASTRONOMY with a space-based experiment ?

XXVIII PHYSICS IN COLLISION - Perugia, June 25-28, 2008

Summary

- 1 Current status of UHECP physics
- 2 UHECP detection from space
- 3 The Super-EUSO mission





Part I

Current status of UHECP physics

Ultra High Energy Cosmic Particles

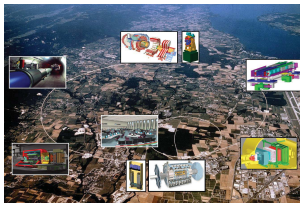
- very low flux:
($\sim 10^{-3}$ particle \cdot km $^{-2}\cdot$ sr $^{-1}\cdot$ yr $^{-1}$
for $E \geq 10^{20}$ eV)
- origin still unknown

If we want to accelerate a proton to such energies using the LHC magnetic field...

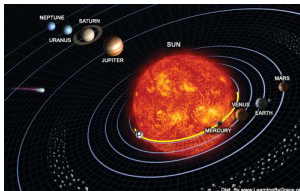


Ultra High Energy Cosmic Particles

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$$B = 8.3 \text{ T}$$
$$R \approx 4.3 \text{ km}$$
$$E = 14 \text{ TeV}$$

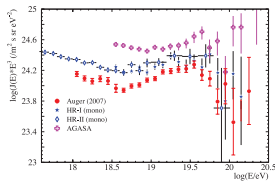


$$B = 8.3 \text{ T}$$
$$E = 10^{20} \text{ eV}$$
$$R \approx 4 \times 10^6 \text{ km}$$

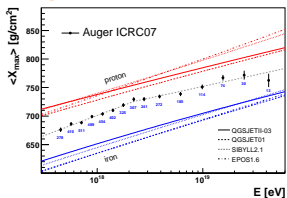
\approx Mercury orbit !

UHECP: what news from the Pierre Auger Observatory ?

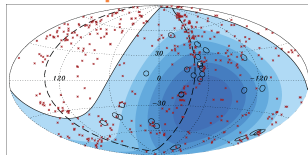
Spectrum



Composition



Anisotropies



see Sergio Petrera's talk

If we can have much more statistics is better! ...



UHECP: what future experiments ?



The next step is the PAO North site, that will be built in the near future.

Together with the South site we will have:

- full sky coverage
- much more event statistics

What if we want to further increase the event statistics ? We need of a large aperture ($\mathcal{A} \sim 10^6 \text{ km}^2 \cdot \text{sr}$).

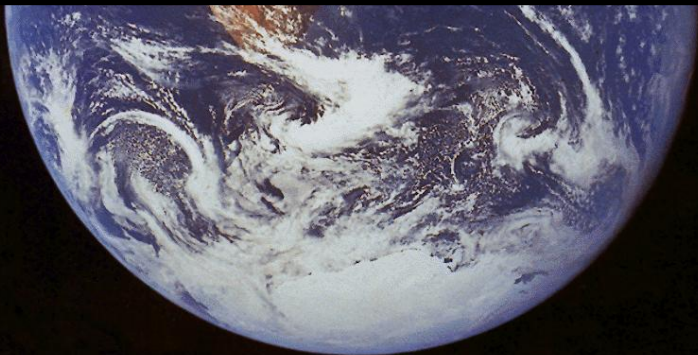
Then a challenging space experiment is probably required.





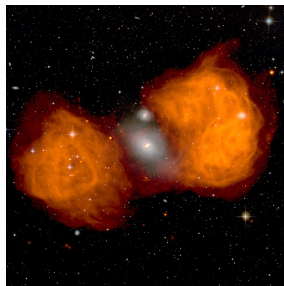
Part II

UHECP detection from space

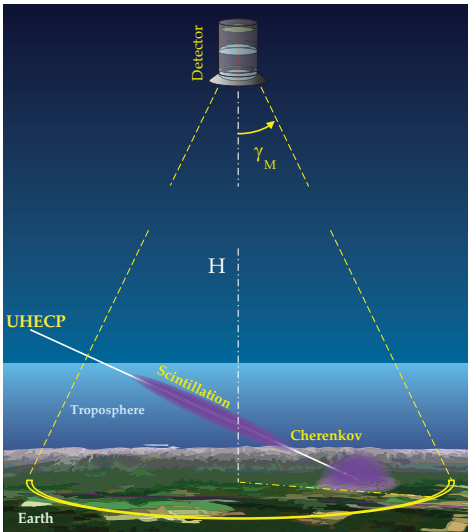


UHECP detection from space: the scientific case

- 1 Extension of the measurement of UHECP spectrum beyond $\sim 10^{20}$ eV
- 2 Detailed map of the UHECP arrival direction, extended to the entire sky
- 3 Identification and localization of compact sources ?
- 4 Study of the spectra of individual sources ?
- 5 Study of the UHECP composition ?



UHECP detection from space: the approach



A large aperture and field of view fast and highly pixelized digital camera.

- Orbital height H
- The apparatus looks downward to the Earth observing
 - near-UV scintillation light
 - diffusely reflected Cherenkov light
- Field of view (FoV) γ_{\max}

The required apparatus is made of:

- main optics (pupil diameter D)
- photo-detector (on the focal surface)
 - photo-sensor (efficiency ϵ)
 - f/e electronics
 - trigger
- atmospheric monitoring system
- calibration system

Performance parameters

- Instantaneous geometrical aperture Goal: $\mathcal{A} \sim 10^6 \text{ km}^2 \cdot \text{sr}$ ($\sim 10 \times \text{PAO}$)
- Energy threshold Goal: $E_{\text{th}} \sim 10^{19} \text{ eV}$
 - superimposition with PAO data
 - high statistics at low energy to study the detector performances
- Energy resolution Goal: $(\Delta E/E)_{\text{stat.}} \sim 10\% + (\Delta E/E)_{\text{sys.}} \sim 10\%$
- Angular resolution Goal: $\sim 1^\circ \div 3^\circ$
- Duty cycle ($\sim 10\% - 20\%$) \rightarrow depends on the background

The above parameters are affected by the **photon collection capability** of the detector. It mainly depends on

- 1 photo-detection efficiency [ε]
- 2 optics aperture [D]
- 3 other efficiency factors (but already close to 1 !)



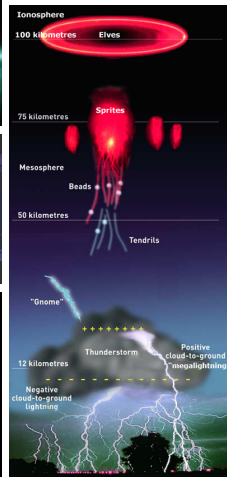
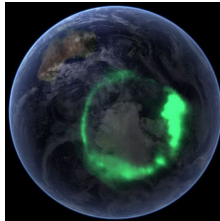
The background from space

Many sources:

- 1 Atmospheric Nightglow
- 2 Man made lights
- 3 Transient luminous events (lightnings,...)
- 4 Auroras
- 5 ...

The mean background flux due to nightglow is $300 \div 1500 \text{ ph}\cdot\text{m}^{-2}\cdot\text{sr}^{-1}\cdot\text{ns}^{-1}$ and increases in presence of clouds and/or moonlight.

A detailed characterization of background in space-time through a microsatellite mission is mandatory to extract the tiny signal from the huge background.



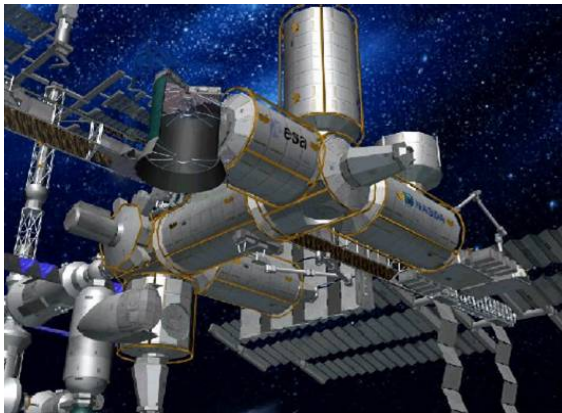


Part III

The Super-EUSO mission

The EUSO experiment

- 1999: proposal as free-flyer satellite
- 2000: proposal for an accomodation on the International Space Station
- 2001: start of Phase A
- 2004: end of Phase A. EUSO is “frozen” due to financial and programmatical issues



The EUSO heritage

- small optics diameter $D_{\text{EUSO}} \sim 2.5$ m
- accommodation on the ISS
- fixed orbit
- strong constraints on mass, volume, power, telemetry
- overall photo-detection efficiency $\varepsilon_{\text{EUSO}} \sim 0.1$
- limited threshold ($E_{\text{th}} \gtrsim 10^{20}$ eV)

The S-EUSO improvement

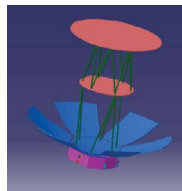
- big optics diameter $D_{\text{S-EUSO}} \sim 7$ m
- free-flyer satellite
- less constraints on the orbit
- less constraints on mass, volume, power, telemetry
- overall photo-detection efficiency $\varepsilon_{\text{S-EUSO}} \gtrsim 0.25$
- better threshold ($E_{\text{th}} \sim 10^{19}$ eV)

EUSO was a preliminary exercise. We have to regard it as a beautiful lesson to build a second-generation UHECP space experiment.

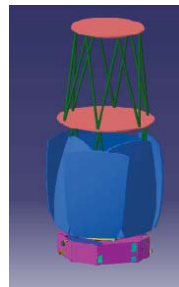
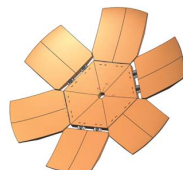
The S-EUSO apparatus (preliminary design)

- catadioptric system (Schmidt telescope)
- lightweight and deployable

mirror diameter	11 m
pupil diameter	7 m
f/#	0.7
granularity at ground	~ 700 m
field of view	$\gamma_{\max} \sim 25^\circ$



in-flight



closed



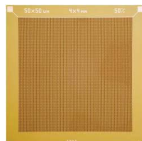
in stowed

The S-EUSO apparatus (preliminary design)

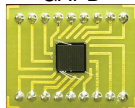
Focal surface

- Array of photo-sensor with high photo-detection efficiency
- For example Geiger-Mode Avalanche Photodiodes (GAPD)

foc. surface diameter	4 m
number of channels	$\sim 10^6$
pixel size	~ 4 mm
photo-detection efficiency	0.25
power per channel	2 mW



4 mm × 4 mm
GAPD



5 mm × 5 mm
GAPD

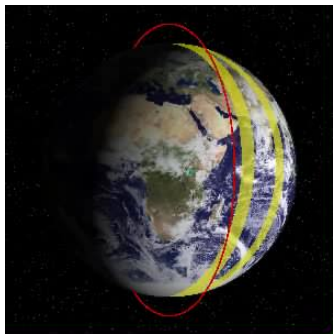


Atmospheric monitoring

- infrared camera (horizontal cloud coverage)
- backscatter LIDAR ??? (cloud-top altitude)

Some remarks about the orbit

- The tuning of the orbit is essential
- A free-flyer satellite has many degrees of freedom
- Possibility to change the orbit during the mission
- E.g. using an elliptic orbit or changing the altitude
 - higher orbit \Rightarrow higher aperture but higher threshold
 - lower orbit \Rightarrow lower threshold but lower aperture
- No tilting with respect to the nadir is required:
 - A small tilting can improve the geometrical aperture
 - But a large tilting has many disadvantages (atmospheric attenuation,..)

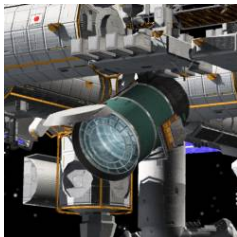


S-EUSO Orbit (preliminary design)

perigee radius	800 km
apogee radius	1100 km
inclination	$50^\circ - 60^\circ$
period	~ 100 min
ground velocity	~ 7.5 km/s

The long road toward S-EUSO

- 1 A microsatellite mission
 - background characterization in space and time
 - validation of the space approach/apparatus using also ground sources
- 2 Pathfinder missions:



JEM-EUSO (Japan)



TUS (Russia)

- 3 Super-EUSO

see also Piero Spillantini's talk

Conclusions

- A space-based detector for UHECP is very challenging
- It is very important to precisely identify the scientific case on the light of the Pierre Auger Observatory results
- The parameters of the apparatus have to be accurately tuned
- The R&D should start as soon as possible
- It is necessary a large effort from the whole UHECP physicist community
- Necessary some preliminary steps (validation, pathfinders,...)
- Possible framework: ESA Cosmic vision 2015-2025 ???



Thank you !



References

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