SEY PROPERTIES IN PLASMA RESEARCH MODELLING AND MEASUREMENTS

Igor D. Kaganovich\textsuperscript{(a)} and Yevgeny Raitses\textsuperscript{(a)}

C. Swanson\textsuperscript{(a)}, V.I. Demidov\textsuperscript{(c)}, A. Mustafaev\textsuperscript{(d)},
A.V. Khrabrov\textsuperscript{(a)}, M. Campanelli\textsuperscript{(a)}, H. Wang\textsuperscript{(a)},
D. Sydorenko\textsuperscript{(b)},

\textsuperscript{(a)}Princeton Plasma Physics Laboratory, Princeton, NJ
\textsuperscript{(b)}Department of Physics, University of Alberta, Canada
\textsuperscript{(c)}University of West Virginia, USA
\textsuperscript{(d)}St. Petersburg Mining University, Russia
Outline

• Introduction
  – PPPL
  – Heavy Ion Fusion Program
  – Plasma applications where SEE is important
• SEY properties: modelling
  – PIC codes; Examples of simulations
• SEY properties: experiments
  – Experimental setups; Example of measurements of SEY
Prof. Lyman Spitzer founded PPPL in 1951 for the Matterhorn Project on magnetic fusion.

In Alps: 4,478 meters (14,692 ft)

Spitzer’s Stellarator

Tokamak experiments

Management, Engineering, DOE office

HTX

Manufacturing

PS&ET

Grad program

Theory

FY2008 Funding: $77.0 million

Number of Employees: 420
- Faculty: 3
- Physicists: 78
- Engineers: 80
- Technicians: 160
- Administrators: 85
- Clerical Support: 14
- Graduate Students: 35

*As of March 1, 2008
Princeton Plasma Physics Laboratory

Tokamak (NSTX)

Stellarator (NCSX) in 2010

LTX

Heavy ion beam

HTX

PFRC/MNX

MRI

MRX
Heavy Ion Fusion Program

• Collaboration between LBNL, LLNL, PPPL
• Goal to produce high intensity ion beam as driver for inertial fusion.

**High Current Experiment (HCX)**
Beam with ~100V self potential
SEY studies for ion beams A. Molvik
Gas Ionization by Ion Beam: I. Kaganovich

**Neutralized Drift Compression Experiment (NDCX)**
Beam Compression in plasma
~70 longitudinal, ~100s times radial
Experiments: P. Roy, P. Seidl
Plasma neutralization theory: I. Kaganovich
Plasma-wall interaction in the presence of strong electron-induced secondary electron emission (SEE)

• Any plasma with electron temperatures above 20 eV for dielectric walls, and above 50-100 eV for metal walls is subject to strong secondary electron emission (SEE) effects:

Hall thrusters and Helicon thrusters
Hollow cathodes for high power microwave electronics
Multipactor breakdown and surface discharges
Space plasmas and dusty plasmas
Fusion plasmas
Plasma processing discharges with RF or DC bias

• Strong secondary electron emission from the floating walls can alter plasma-wall interaction and change plasma properties.

• Strong SEE can significantly increase electron heat flux from plasma to the wall leading to: 1) wall heating and evaporation and 2) plasma cooling.
Plasma applications where SEE is important

Hall Thruster discharge: used for electric propulsion

Magnetron discharge: used for deposition, plasma switch for electric grid

From: www.angstromsciences.com
Modelling of SEE in plasma research

- 3D BEST PIC code: includes electromagnetic (Darwin scheme) and electrostatic modules.  
  https://nonneutral.pppl.gov/

- 3D LSP code includes electromagnetic and electrostatic modules. In collaboration with Voss Scientific.

- 1-2D PIC code EDIPIC. Implemented electron-atom scattering, ionization, and excitation as well as electron-ion and electron-electron collisions, complex SEY models.  
  https://w3.pppl.gov/~ikaganov

SEE electron effects on sheath and electron energy distribution functions 5 recent PRLs

two-stream instability

electrons during neutralization

SEE electron effects on sheath and electron energy distribution functions 5 recent PRLs
Plasma properties can be changed by applying engineered materials to the surface.

Application of carbon velvet to channel walls improves considerably thruster performance by reducing the electron cross-field current and by increasing nearly twice the maximum electric field in the channel compared with the conventional BN ceramic walls.

- Velvet suppresses SEE and reduces current at high voltages (good)
- Sharp tips can enhance field emission leading to arcing (bad)
- Need to engineer velvet morphology so that inter fiber gaps and protrusions are located well inside the sheath to avoid damage by arcing
- Need to take into account spatial and temporal variations of sheath width due to plasma non-uniformity or instabilities
Simulations and Theory of SEY of complex surfaces: Velvet

Velvet: regular or irregular lattice of normally-oriented fibers

\[ u = \frac{\pi}{2} DA = 2rnh \]

\( u \) dimensionless parameter, \( D \) area packing fraction, \( A \) aspect ratio of fibers, \( r \) radius of fibers, \( n \) area density of fibers, \( h \) height of fiber layer

SEY as a function of incident angle for different packing density of velvet.

Lines: Analytic model.
Points: Monte-Carlo simulations.

Discrepancy is due to tertiary and higher-order electrons.

Velvet is well-suited to suppressing normally incident primary electrons

Simulations and Theory of SEY of complex surfaces: Fuzz/foam

Fuzz/foam: irregular lattice of isotropically-oriented fibers

\[
u = \frac{\pi}{2} DA = 2rnh\]

\(u\) dimensionless parameter, \(D\) area packing fraction, \(A\) aspect ratio of fibers, \(r\) radius of fibers, \(n\) area density of fibers, \(h\) height of fiber layer

SEY as a function of incident angle for different packing density of foam.
Lines: Analytic model.
Points: Monte-Carlo simulations.
Discrepancy is due to tertiary and higher-order electrons.

Foam is not well-suited to suppressing normally incident primary electrons

Simulations and Theory of SEY of complex surfaces: Feathers

Feather: lattice of normally-oriented fibers with smaller, secondary fibers on the sides of that fiber.

SEY as a function of incident angle for different packing density of foam. Feathers are able to suppress SEE for all electrons $\sim 1/4$.

PPPL experimental setup for SEE measurements

- 10^{-8} - 10^{-10} \text{Torr (turbo, ion, & Ti sublimation pumps)}
- Quadrupole Mass Spectrometer
  - Background gas, temperature program desorption (TPD)
- Kimball Physics Pulsed Electron Source
  - SEE measurements of dielectric and conductive materials
- Auger Electron Spectroscopy (AES)
  - Sample composition, SEE
- Low Energy Electron Diffraction (LEED)/AES
  - SEE yield, angular dependence and energy distribution of SEE electrons
- Electron Cyclotron Resonance Plasma Source
  - Sample cleaning
- Resistive heating (~1400K max)
  - Sample cleaning & conditioning, TPD
- LN\textsubscript{2} cooling (<200K)
- High Resolution Electron Energy Loss Spectroscopy
- X-ray Photoelectron Spectroscopy (XPS)
PU experimental setup incorporates *in situ* analysis of material composition.

- **Vacuum Chamber**
  - Auger Electron Spectrometer
  - O2 gas line
  - Sample
  - P=2x10^{-10} Torr
  - Ni(110) substrate
  - 40 ML (10 nm) of lithium on surface
  - Can expose to O\textsubscript{2} & H\textsubscript{2}O to adjust composition
  - Measure composition with Auger electron spectroscopy
Upgraded setup for measurements of SEE yield from micro-engineered materials

- Cryogenic system to maintain better vacuum (<10^{-8} torr) during SEE measurements
- Ion source to remove surface charges
- The upgrade allows to minimize, outgassing, surface, contamination, etc.
SEY of surface micro-architected engineered materials to suppress SEE

- Surface-architectured materials can reduce the effective SEE yield by trapping SEE electrons between surface architectural features.
- The SEE reduction is more significant for high aspect ratio (1:10^3) velvets than for low aspect ratio (1:10) dendritic coatings.

**Measured total SEE yield from velvets and dendrites:**

- Carbon velvet
- Dendrites
- Mo-coated Re
- W-coated Re

![Graph showing SEE yield vs. Energy of primary electrons, eV]

- BN M26, 400 C
- BN M26, RT
- 10% W-Foam
- 10% SiC-Foam
- Graphite
- Carbon Velvet
Measured electron energy distribution function (EEDF) for true SEE – important input for plasma-wall interaction.
SEE yield for W-flat samples

Fig. Total SEE yield of smooth pre-sputtered (blue asterisks) and post-sputtered (black filled circles/triangles) W at 0° and 45°.

SEE from post-sputtered W at 0° matches previous results of cleaned W (green and purple lines), and at 45° follows a $1/\cos(\Theta)$ dependence (black unfilled circles).

SEE from pre-sputtered W is higher than from post-sputtered W since C, O, and many oxides increase SEE.
SEE yield for W-fuzz and W-flat samples
Angular dependence of SEE yield

**Fig:** Total SEE yield from W fuzz at 0° and 45° (red squares/crosses) compared to smooth post-sputtered W (black circles/triangles). SEE from W fuzz is >40% lower than from smooth W (despite W fuzz having more C, O, oxidation) due to trapping of secondary electrons within the fuzz. SEE from W fuzz is independent of primary electron incident angle since the orientation of fibers leads to a wide distribution of local incident angles.

**Fig:** SEM image of (top) top view and (bottom) side view of W fuzz formed when exposed to He plasma at elevated temperatures (i.e., 60 eV He\(^+\), flux = 3.7x10\(^{21}\) m\(^{-2}\)s\(^{-1}\), fluence = 1.3x10\(^{25}\) m\(^{-2}\), sample at 1270 K). Fibers are 25-50 nm in diameter and 100-200 nm long.
**Velvet: surface-architectured material with low SEE**

- Total SEE yield at normal incidence measured in vacuum.

- SEE from velvet can be several times lower than SEE from carbon.

*Jin, Ottaviano, Raitses (2017)*
As oxygen content increases, SEY greatly increased.

Water is major contaminant in vacuum systems and gives yields similar to fully oxidized lithium.

Yield of LiOH similar to oxidized Li

SEY was measured within few eV precision using a wall probe using penning produced electrons with a specific energy.

A metallic boundary reflects a *negligible amount* of low-energy incident electrons when uncontaminated ("clean") and reflects a significant amount when contaminated by monolayers of adsorbent.

\[
A^* + A^* \rightarrow A + A^+ + e
\]

Penning ionization gives electrons of fixed energy.

---

It is shown that for poly-crystal surfaces, the SEE yield can be indeed very high (~0.8) but still not approaching unity. This result is explained by additional reflection of primary electrons from a potential barrier near the poly-crystal surface. The contribution of electron reflection from the potential barrier and the surface has been indentified and studied.

A. Mustafaev, et al., to be submitted (2018).
Conclusions

• Derived analytical formulas for Secondary Electron Emission Yield for complex surfaces: velvet, foam/fuzz, feathers and verified with a MC code.

Feathered surfaces are best at reducing SEY by a factor of 4.

• PPPL has sophisticated experimental set ups to measure SEY in cleaned and oxidized samples, including dielectrics.

Measured SEY for several surface micro-structured engineered materials to suppress SEE: velvet, fuzz, dendritic coatings.

High-aspect-ratio velvet reduces SEY most compared to low-aspect-ratio dendritic coatings.

Measured EEDFs of true secondaries at low energies.

Measured angular dependence on primary electrons of SEY for W and fuzz.

Measured effect of oxidation on SEY of W, W fuzz and Li.

Measured SEY of very low energy using penning reaction in plasma magnetized thermionic discharge.
PPPL Relevant References


Relevant Bibliography
Secondary electron emission yield from dielectric materials

Note:
for Boron Nitride ceramic, if plasma (primary) electrons have Maxwellian electron energy distribution function (EEDF):

\[ \gamma(T_e) = 1 \text{ at } T_e = 18.3 \text{ eV} \]

“Sample” method to measure SEE yield from dielectric and high electric resistance materials

- Faraday cup to measure the primary electron current, $I_p$
- Sample to ground current to measure the sample current, $I_s$
- A slightly positively biased collector to attract SEE electrons
- SEE current is obtained from $I_{SEE} = I_p - I_s$
- SEE yield is estimated as $\gamma = I_{SEE} / I_p$
Measurements of SEE Properties of Materials

- PPPL Electron LEED-Auger Spectroscopy System:
  - UHV facility: $1 \times 10^{-8}$ Torr.
  - Thermionic emission electron gun: 3-1600eV.
    - Retarding potential analyzer for measurements of EEDF of SEE electrons.
    - Conducting and dielectric materials.
- Use two measurement methods of the SEE yield:
  i) biased sample
  ii) biased collector

Energy level diagram for LEED/AES optics.
High signal-to-noise measurements of SEE currents

- Fast amplifiers with bandwidth of 10 MHz, gain $>10^7$ V/A (1 V for 100 nA) and the current resolution of <1 nA.
- Reference method – Faraday cup signal is subtracted from the Sample signal to compensate for ambient noise during the pulse.
- Example of the measured sample current from 95% Al$_2$O$_3$

Primary energy: 300 eV
Sample current: 20 nA
Pulse: ~ 5 µs
SEE yield: $\gamma > 1$
SEE Properties of Ceramic Materials and Graphite

- Strong SEE effects on plasma-wall interaction occur when SEE approaches 1.
- For ceramic materials, SEE yield is higher and approaches 1 at lower energies than for metals due to a weaker scattering of SEE electrons on phonons (for insulators), $\lambda \sim 20$ nm, than on electrons (for metals,), $\lambda \sim 1$ nm.
Fuzz characterization

Fig 6: Front view of the facility showing the W fuzz sample under the X-ray source.
Fig 8: XPS spectra of smooth pre-sputtered W (dashed blue line), smooth post-sputtered W (thin black line), and W fuzz (thick red line). W fuzz has WO$_x$ and more C and O impurities than the smooth W samples (full XPS spectra not shown).
Plasma properties can be changed by applying engineered materials to the plasma facing surface.

Application of high aspect ratio carbon velvet to thruster channel walls improves considerably thruster performance by reducing the electron cross-field current and by increasing nearly twice the maximum electric field in the channel compared with the conventional BN ceramic walls.

2 kW Hall thruster
12 cm channel OD

Carbon velvet fibers:
Diameter $\approx 5 \, \mu$, $L \approx 2000 \, \mu$, $g \approx 20 \, \mu$

Velvet suppresses SEE and reduces electron cross-field current as compared to other materials.
Effect of anode material on the breakdown in low-pressure helium gas

To demonstrate the effect of the anode material on the breakdown in low-pressure helium gas, systematic experiments in helium were conducted using the copper cathode and a variety of materials for the anode. A wineglass discharge tube shown in the left figure was used. Results of measurements of the left sides of the Paschen curves are shown in the middle figure. The curve for graphite is substantially shifted to the right. The right figure demonstrates multi-value breakdown points for the graphite anode accessed by (1) increasing and (2) decreasing the applied voltage.