



# ESA Perspectives on Electric Propulsion

Käthe Dannenmayer

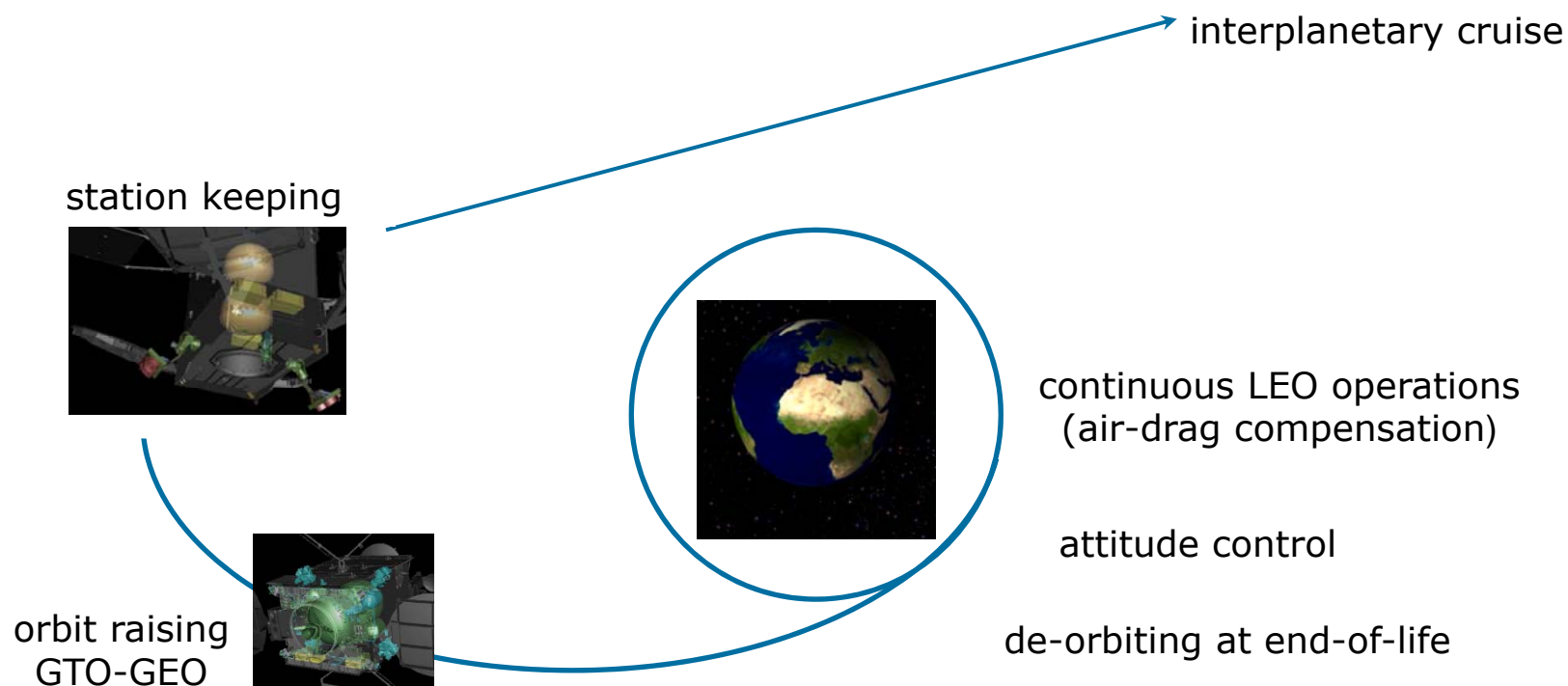
02/03/2017

ESA UNCLASSIFIED - For Official Use



European Space Agency

# Electric Propulsion - Applications

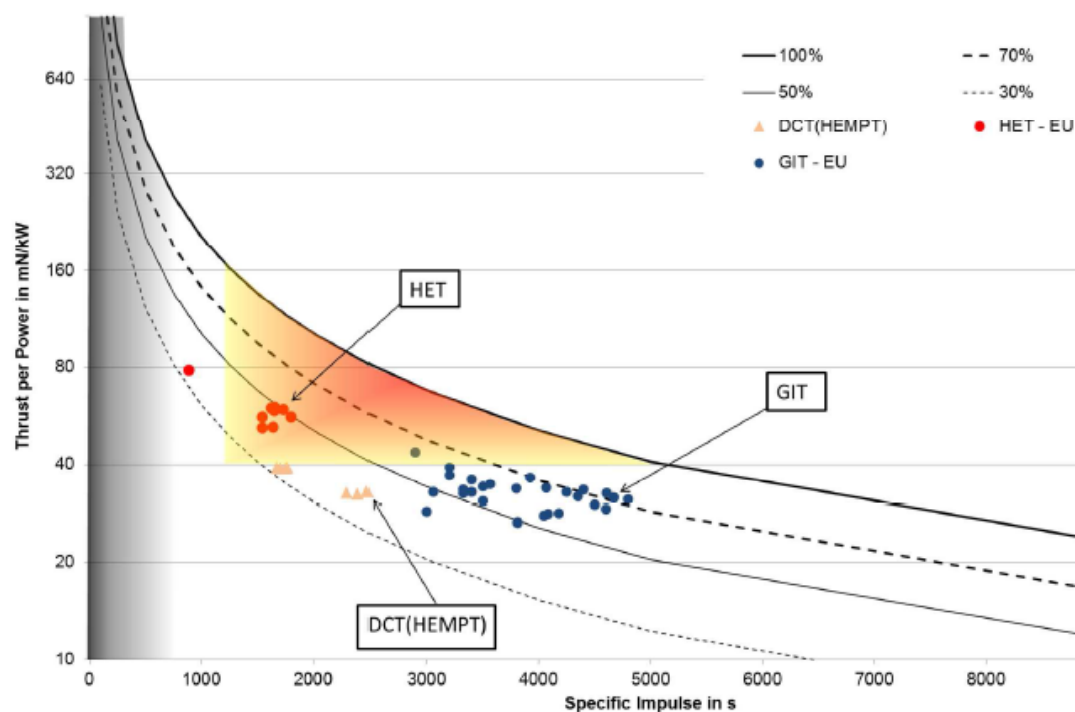


# Electric Propulsion - Applications



Required Performance for Orbit Raising of MEO & GEO Missions

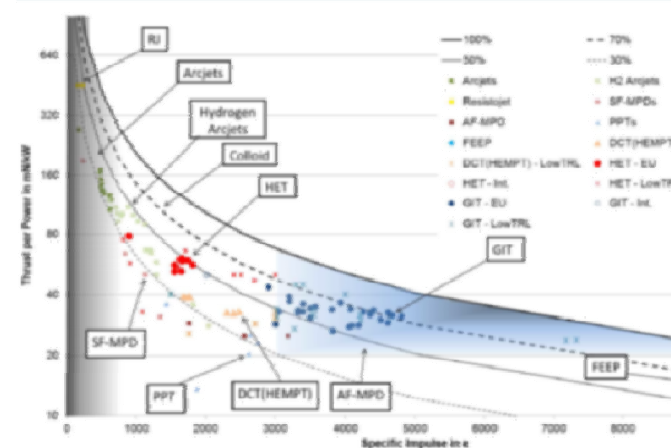
European Thrusters (high TRL only)



ESA UNCLASSIFIED - For Official Use

Required Performance for Deep Space Missions

European Thrusters including low TRL



Käthe Dannenmayer | IPAIA2017 | 02/03/2017 | Slide 3



European Space Agency

# Electric Propulsion - Applications

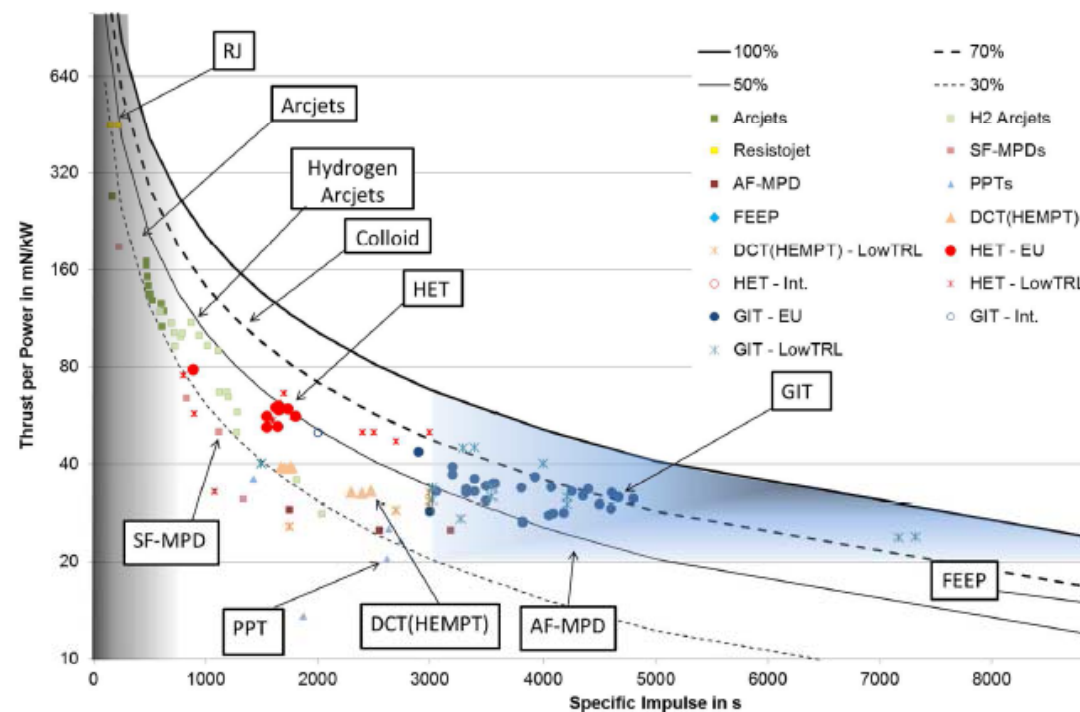
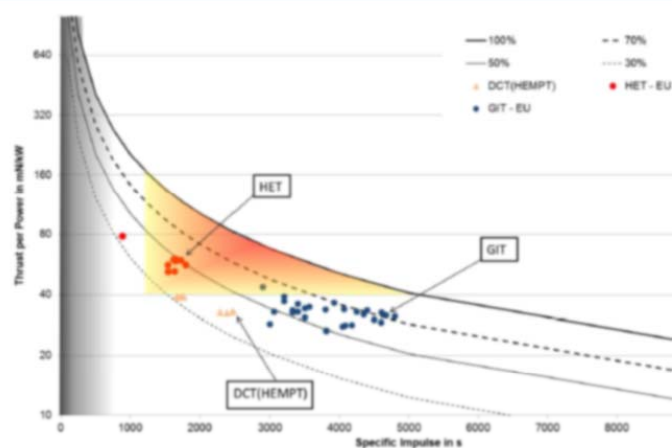


## Required Performance for Deep Space Missions

### European Thrusters including low TRL

#### Required Performance for Orbit Raising of MEO & GEO Missions

##### European Thrusters (high TRL only)



ESA UNCLASSIFIED - For Official Use

Käthe Dannenmayer | IPAIA2017 | 02/03/2017 | Slide 4



European Space Agency

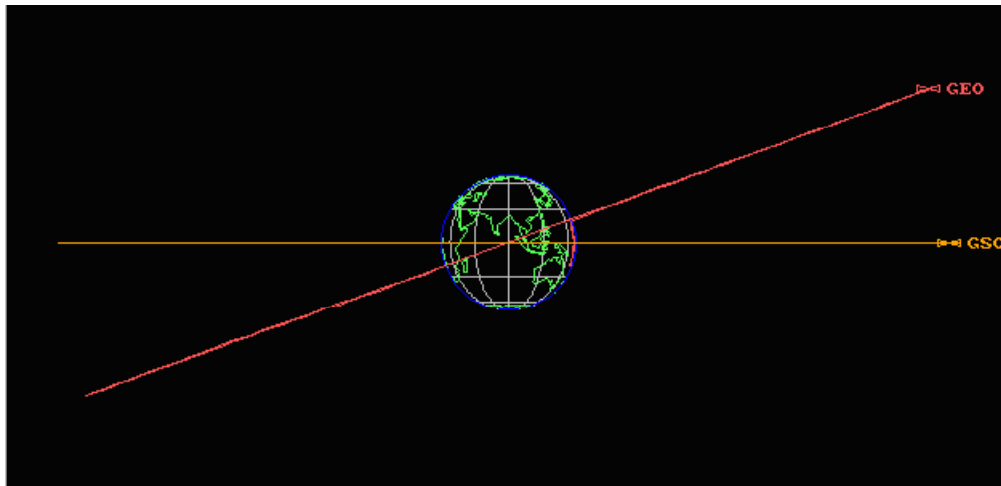
# Electric Propulsion - Applications

- EP is unlikely to ever completely replace chemical
  - Low thrust  $\equiv$  long manoeuvre durations
  - EP cannot be used to bring a spacecraft into space
- At present there are two principal applications for EP
  - **Commercial/GEO satellites**
    - Station-keeping (NSSK & EWSK)
    - Orbit raising
    - Orbital re-positioning and end-of-life disposal
  - **Scientific satellites**
    - Near Earth science missions e.g. drag compensation at low altitude
    - Primary propulsion for deep space missions
    - Fine pointing/station-keeping

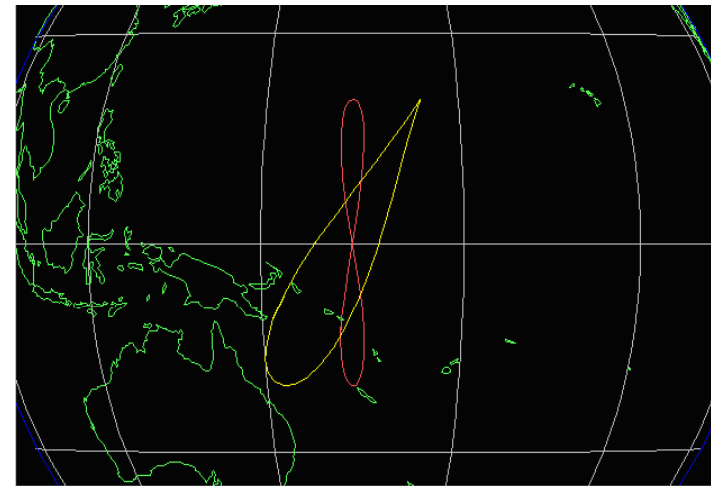
# Telecommunication - NSSK



- Station-keeping propulsion is required to correct for orbital perturbations
  - Sun - gravitational and solar radiation pressure
  - Moon - gravitational
  - Earth oblateness



Geostationary Orbital Plane - side view

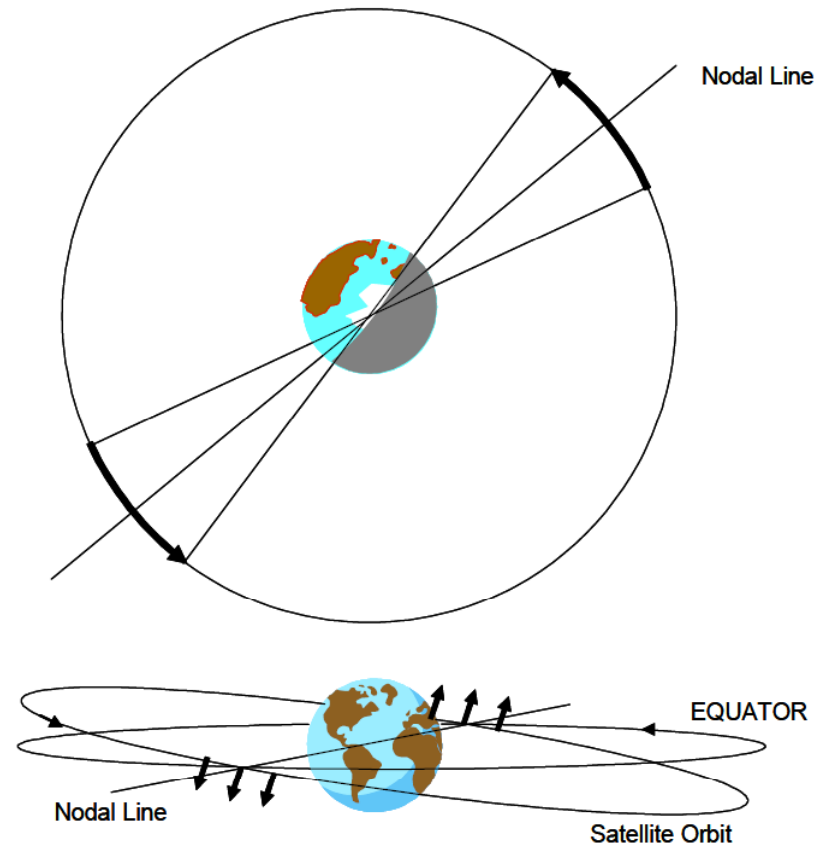


Ground Track

# Telecommunication - NSSK



- If left uncorrected, these effects will increase the orbit inclination by up to  $1^\circ$  per year compared with a typical operational need to maintain inclination to within  $0.05^\circ$
- Correction manoeuvres (station-keeping) performed around orbital nodes
- Thrusters fired in North direction and then in South direction on opposite side of orbit
- NSSK accounts for  $\sim 50 \text{ms}^{-1} \Delta V$  per year



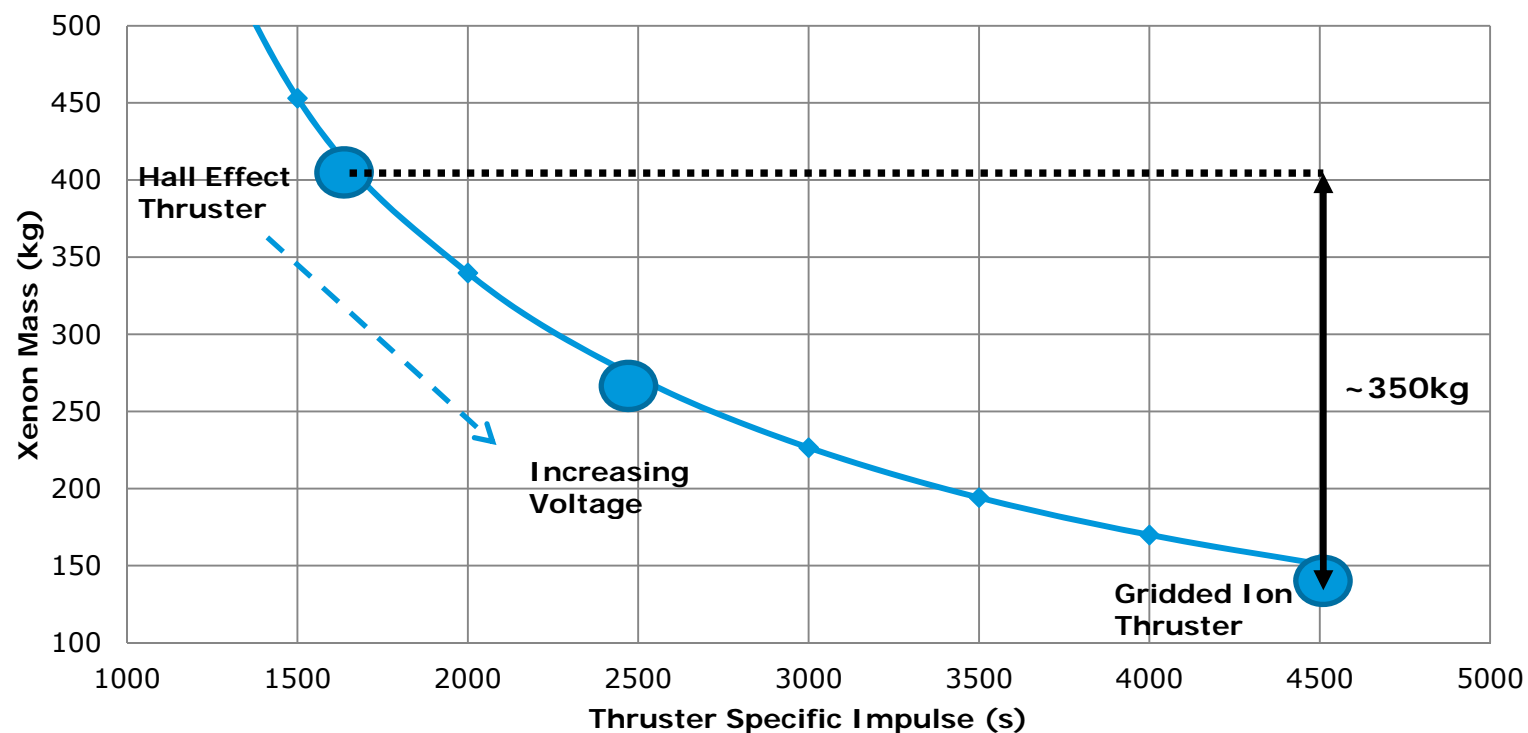
ESA UNCLASSIFIED - For Official Use

Käthe Dannenmayer | IPAIA2017 | 02/03/2017 | Slide 7



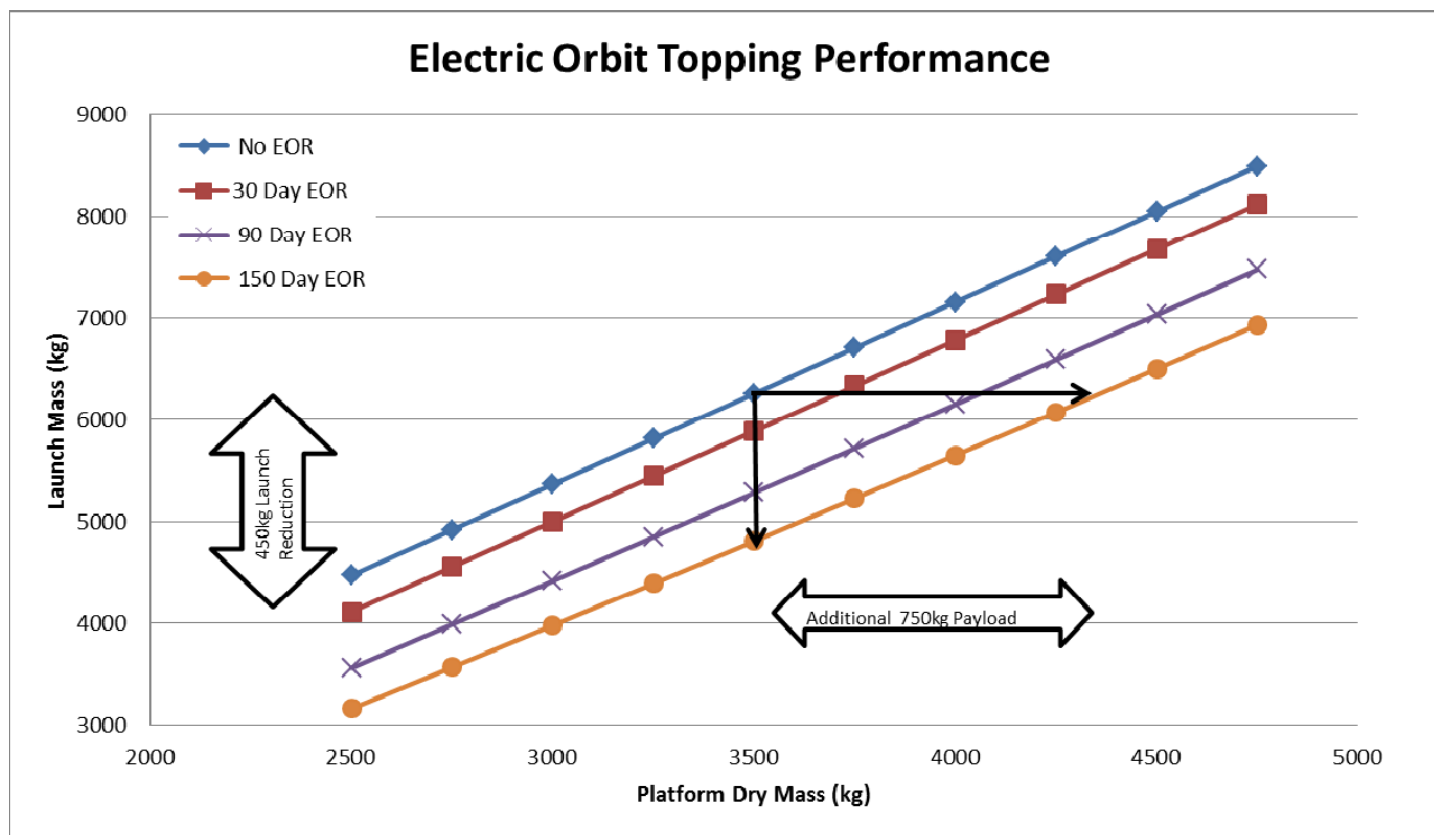
European Space Agency

## Station Keeping Xenon Mass versus Thruster Isp





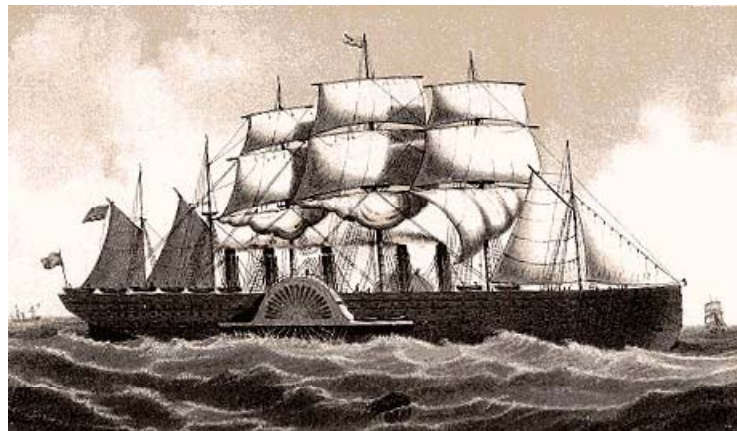
# Telecommunication – Orbit Topping/Raising



# Telecommunication



- For thousands of years people relied on sail – they knew how to use it and understood its capabilities and drawbacks – 'better the devil you know'
- Steam power offered a quantum leap in performance - but the systems were initially unproven and operators were nervous
- As a consequence hybrid ships – sail & steam - were built and used until confidence in steam systems increased to the point that business was confident enough to remove sails altogether – **EP is at this turning point**

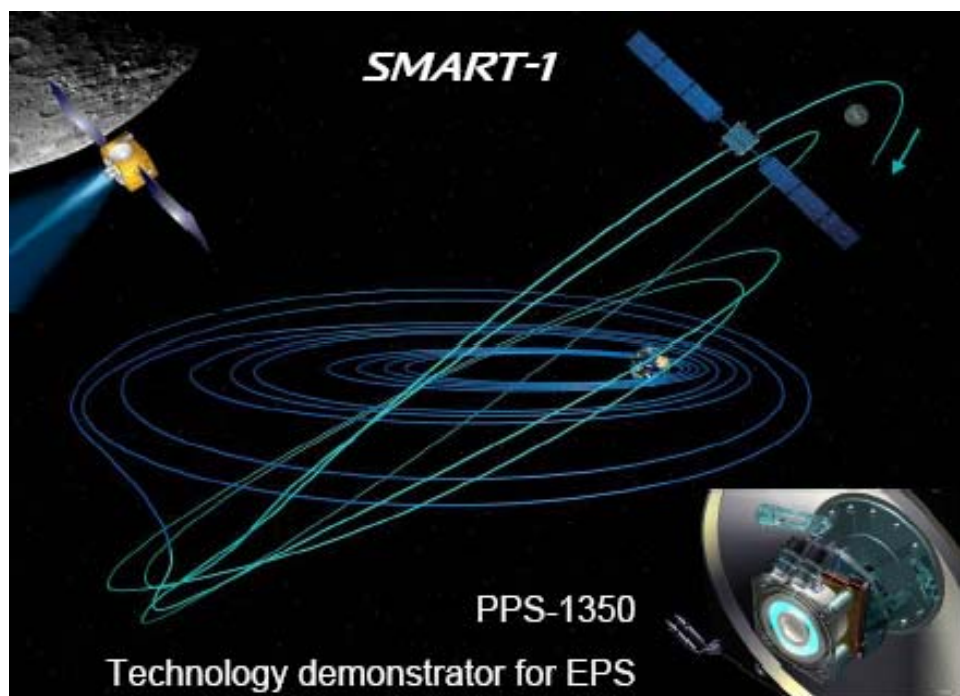


## Navigation

- ESA is preparing the future replacement of GALILEO constellation and is targeting the possibility to increase the Galileo Payload capability without impacting the launch costs
- The increase in payload capability could be achieved by changing the launch injection strategy and by using EP to transfer the satellite to the target operational orbit.
- The use of EP might allow to use small launchers such as VEGA or place more spacecraft in the current SOYUZ and Ariane 5 launchers.
- GIE and HET subsystems are currently considered for the transfer by the selected Primes of Phase A/B1.



# Science and Earth Observation – SMART1



- Mass 370 kg, Size 1 m<sup>3</sup>
- 1.9 kW solar power
- 1.5 kW HET (PPS®1350)
- 16 Months transfer to Moon
- 80 kg of Xe
- Launch: September 2003



ESA UNCLASSIFIED - For Official Use

Käthe Dannenmayer | IPAIA2017 | 02/03/2017 | Slide 12

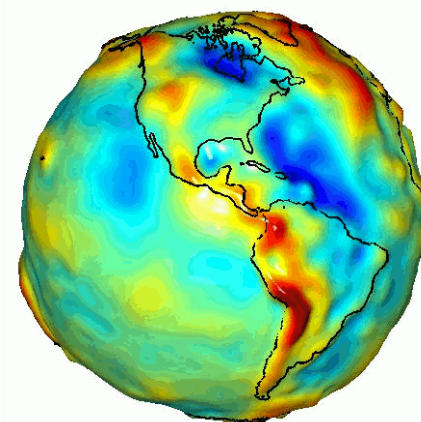


European Space Agency

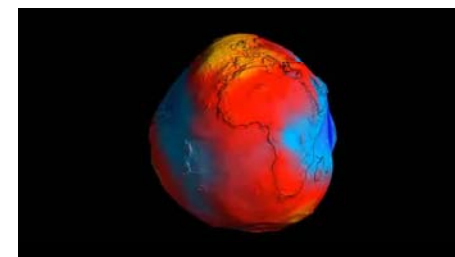
# Science and Earth Observation – GOCE



- Mass: 1050 kg
- Size: 5 m long, 1m<sup>2</sup> cross section
- 1.3 kW solar power
- Two 20 mN GIE (T5) for drag compensation
- Launch: March 2009



*Computer model (exaggerated) of equipotential reference surface*



*GOCE science data*

ESA UNCLASSIFIED - For Official Use

Käthe Dannenmayer | IPAIA2017 | 02/03/2017 | Slide 13

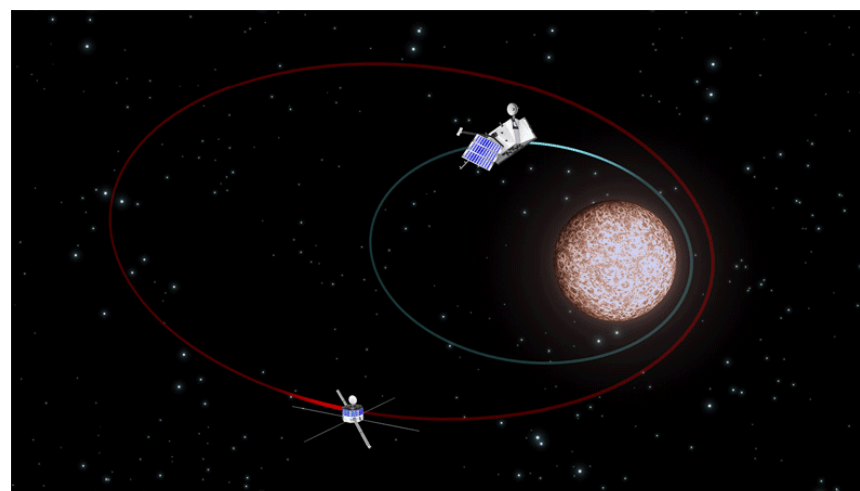
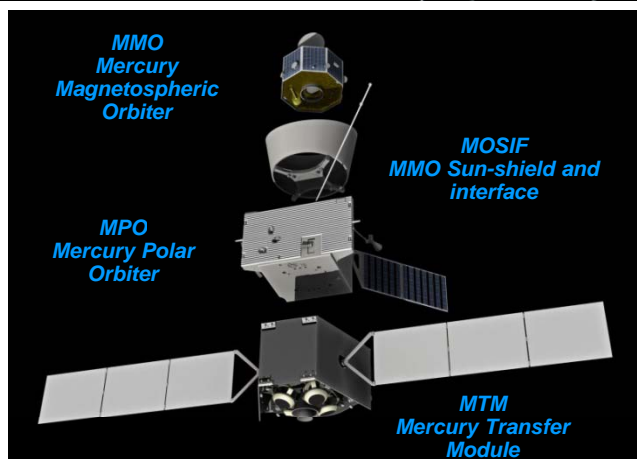


European Space Agency

# Science and Earth Observation – BepiColombo



- Mass: 4100 kg
- 2 GIE (T6), 290 mN total thrust
- 7.5 years long transfer to Mercury using EP
- Launch: October 2018 (planned)

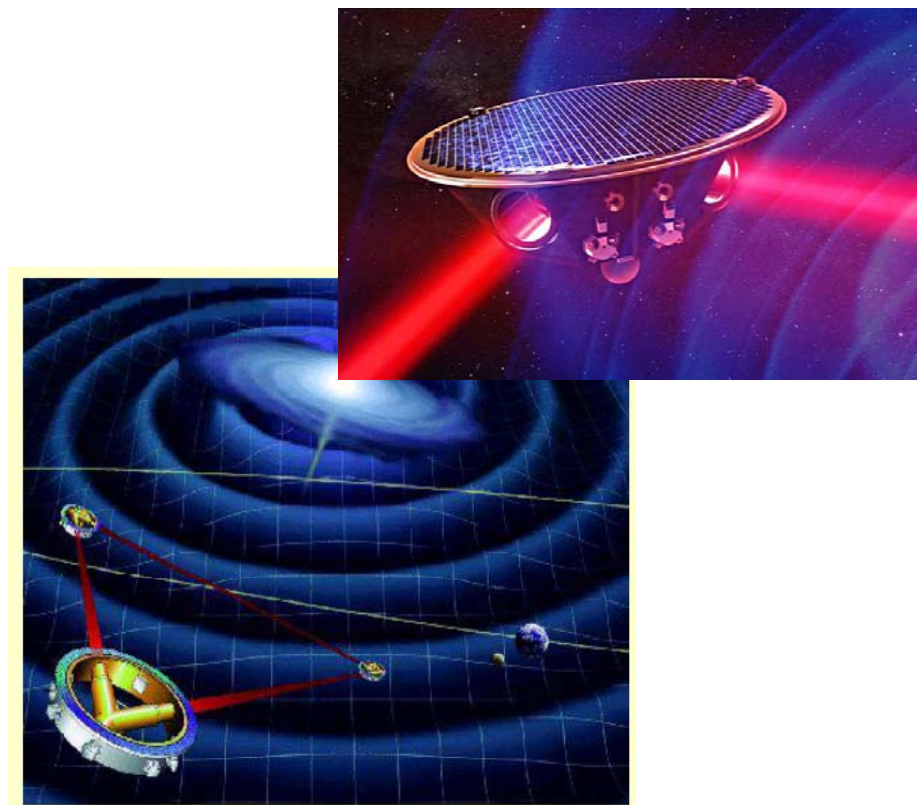




## Science and Earth Observation – LISA



- Primary objective: detect and observe gravitational waves
- Mission success based on performance of sophisticated accelerometers working under drag-free conditions on each LISA spacecraft
- 3 clusters of four micro thrusters on each LISA spacecraft ( $\sim 50 \mu\text{N}$  each)
- LISA Pathfinder technology demonstrator mission launched in December 2015



## Science and Earth Observation – NGGM



- Follow-up mission of GOCE
- Primary objective: measure variations of Earth's gravity field over long time span (possible covering full solar cycle)
- Measurement technique: "Satellite-to-satellite tracking" (SST) in low Earth orbit
- Pair of satellites flying in loose formation to form "gravity sensor"





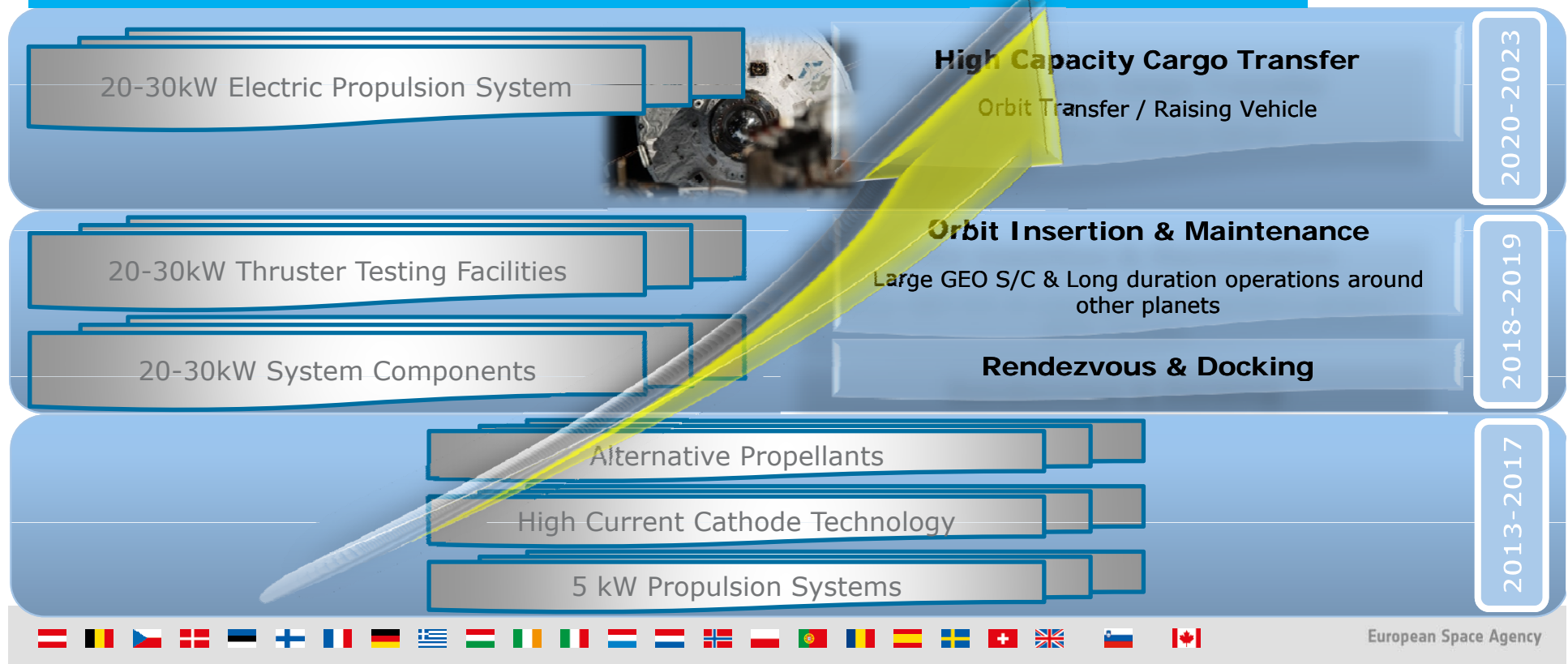
# Electric Propulsion - Outlook



## Exploration:

Application Area: Advanced Propulsion *(Priority for Space Council)*

Technology Subject: *Electric Propulsion for High Capacity Cargo Transfer*



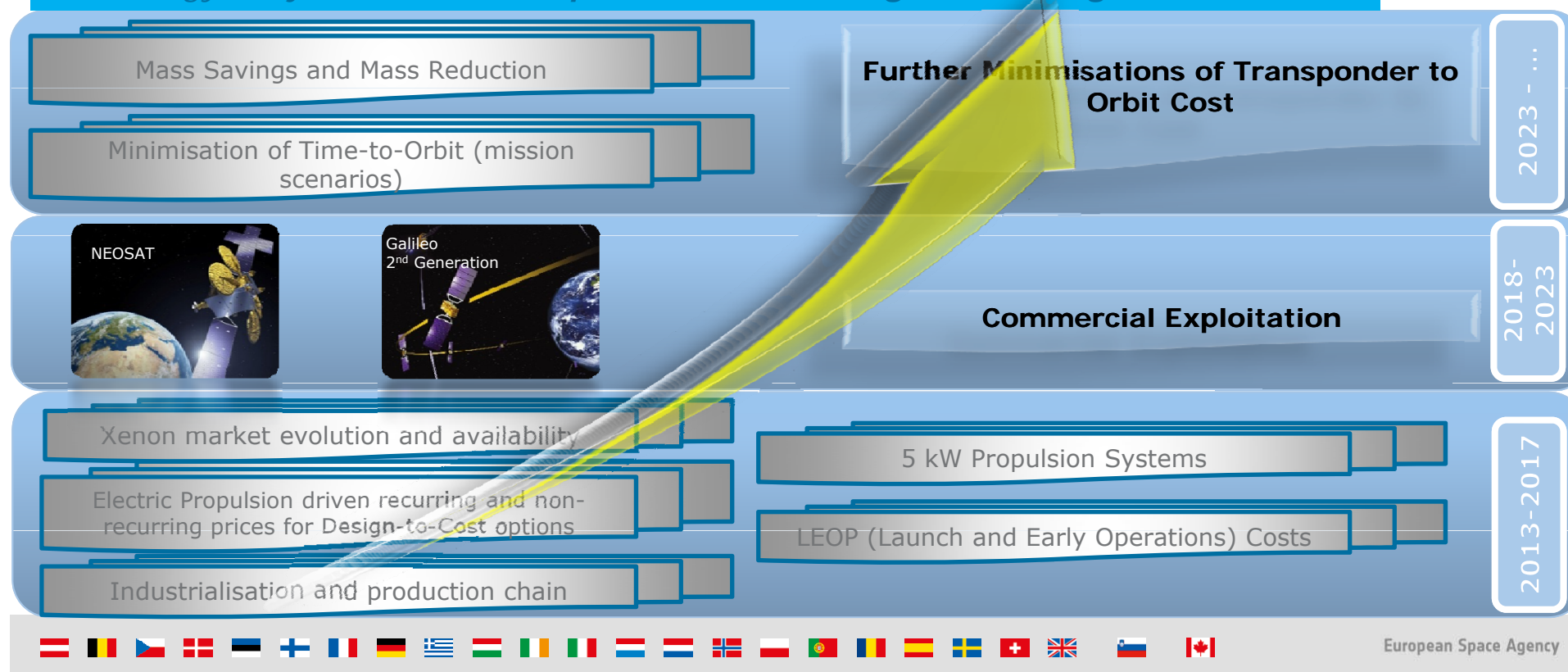
# Electric Propulsion - Outlook



## Commercial Applications:

Application Area: Telecom, Navigation

Technology Subject: *Electric Propulsion for Orbiting, Deorbiting, Orbit Control*

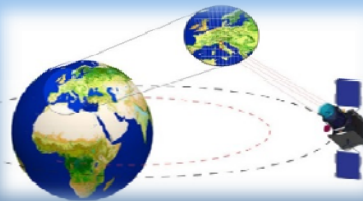


# Electric Propulsion - Outlook



## EO and Mega Constellations:

*Technology Subject: Electric Propulsion for Orbiting, Deorbiting, Orbit Control, Fine Attitude Control, Drag Compensation*



**High Resolution from GEO (EO)**

**Next Generation Gravity Mission (EO)**

2020 - 2024

### Propulsion Subsystem:

- Low cost (constellation)
- Low Power Consumption (EO and Constellation)
- Light (EO and Constellation)
- High thrust controllability (EO)
- Manufacturing Capability (Constellation)



2018-2020

Alternative Propellants for Mini, Micro, Nano Satellites

Power

Micro/Mini Propulsion

### Nano/Microsatellite Applications

Scientific Research, Technology, EO, Education, Military, Astronomy

2013-2017



European Space Agency

# Horizon 2020 Space Work Programme 2014 - EPIC



- Electric Propulsion has been identified as strategic technology by European actors
- European Commission has set up “In-space Electrical Propulsion and Station-Keeping” Strategic Research Cluster (SRC) in Horizon 2020 to enable European leadership of European capabilities in electric propulsion at world level within 2020-2030
- Electric Propulsion Innovation & Competitiveness (EPIC) project aims at providing integrated roadmap of activities and master plan for coordination and implementation through SRC
- EPIC Programme Support Activity (PSA) coordinates activities within SRC, monitors evolution of works performed within SRC and supports commission on implementation
- PSA coordinator: ESA
- PSA partners: ASI (Italy), CDTI (Spain), CNES (France), DLR (Germany), Eurospace, SME4Space VZW, BELSPO (Belgium), UKSA (UK)

## EPIC – Selected Operational Grants

- 6 Operational Grants (OG) selected with a total budget of ~28 M€
- 3 OGs have been awarded under COMPET-3-2016-a (Incremental Technologies)
  1. CHEOPS: focused on Hall effect thrusters
  2. HEMPT-NG: focused on HEMP thrusters
  3. GIESEPP: focused on gridded ion engines
- 3 OGs have been awarded under COMPET-3-2016-b (Disruptive Technologies)
  1. GaNOMIC: focused on a disruptive power converter
  2. HiperLoc-EP: focused on Electrospray Colloid Thrusters
  3. MINOTOR: focused on Electron Cyclotron Resonance Accelerator

<http://epic-src.eu/>

# ESA Activities in the Field of Electric Propulsion



- Technical support to ESA project
- Coordination of R&D activities in the field of Electric Propulsion to prepare for future mission needs
- Coordination of numerical and experimental works in the field of plasma – spacecraft interactions induced by Electric propulsion with the aim to provide improved modelling tools
- Working group on “Standardization of Electric Propulsion Testing and Qualification” in order to achieve cost reduction by means of optimised production and verification programs
- Inter-laboratory comparison of Electric Propulsion Devices

# ESA Propulsion Laboratory



- Located at ESTEC, The Netherlands
- The laboratory is focused on space propulsion
- EPL provides test services to the Propulsion and Aerothermodynamics division, which is responsible for R&D activities and support to projects in the areas of chemical propulsion, electric and advanced propulsion, and aerothermodynamics.
- ISO 17025 Accreditation
  - Thrust, mass flow and electrical power measurements
  - Electric, chemical and cold gas propulsion systems
- Flexible scope of Accreditation:
  - Force: 1  $\mu$ N – 500 mN
  - Mass flow 1  $\mu$ g/s – 300 mg/s
  - Power: 1 mW – 2 kW

## Conclusion

- Electric Propulsion in Europe has been successfully used on missions for Telecom, Science and Earth Observation
- 5 kW engines will be used in the near term on telecommunication satellites for orbit control and full or partial orbit transfer
- Navigation, Science (interplanetary missions) and Exploration (the Moon, Asteroids and Mars) will in the future also require 5 kW systems
- ESA is coordinating development of micro thrusters such as mini-ion engines, FEEPs, mini-Halls, etc. with capability to fulfil stringent Science and Earth Observation requirements
- Galileo 2nd Generation program is considering the use of electric propulsion to perform orbit raising from LEO/GTO to MEO
- In the future developments have to aim for longer lifetime, improved power-to-thrust ratio and higher specific impulse
- Thruster in the 10 to 20 kW range will have to be developed
- Constellations of satellites may make use of EP systems at very low cost, main application will be for end-of-life de-orbiting
- In the frame of the Horizon 2020, the EPIC project has been selected to improve the competitiveness of European actors in the field of electric propulsion