

Improvements in global modeling of gridded ion thrusters

R. Lucken, T. Lafleur, P. Grondein, A. Bourdon, P. Chabert, A. Aanesland



Laboratoire de Physique des Plasmas

LPP, CNRS, Ecole polytechnique, UPMC Univ Paris 06, Univ. Paris-Sud, Observatoire de Paris, Université Paris-Saclay, Sorbonne Université, PSL Research University, 91128 Palaiseau, France

Abstract

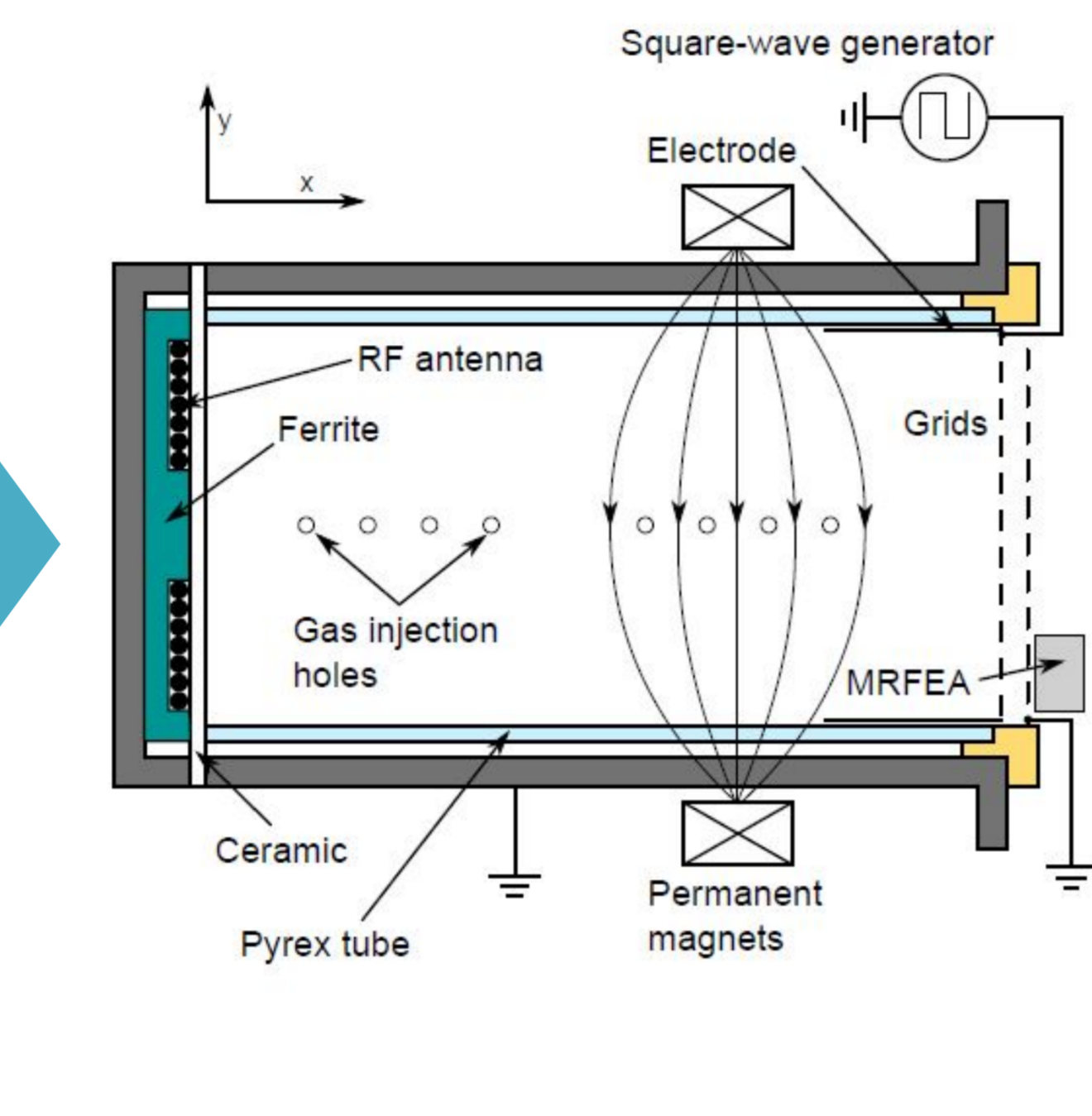
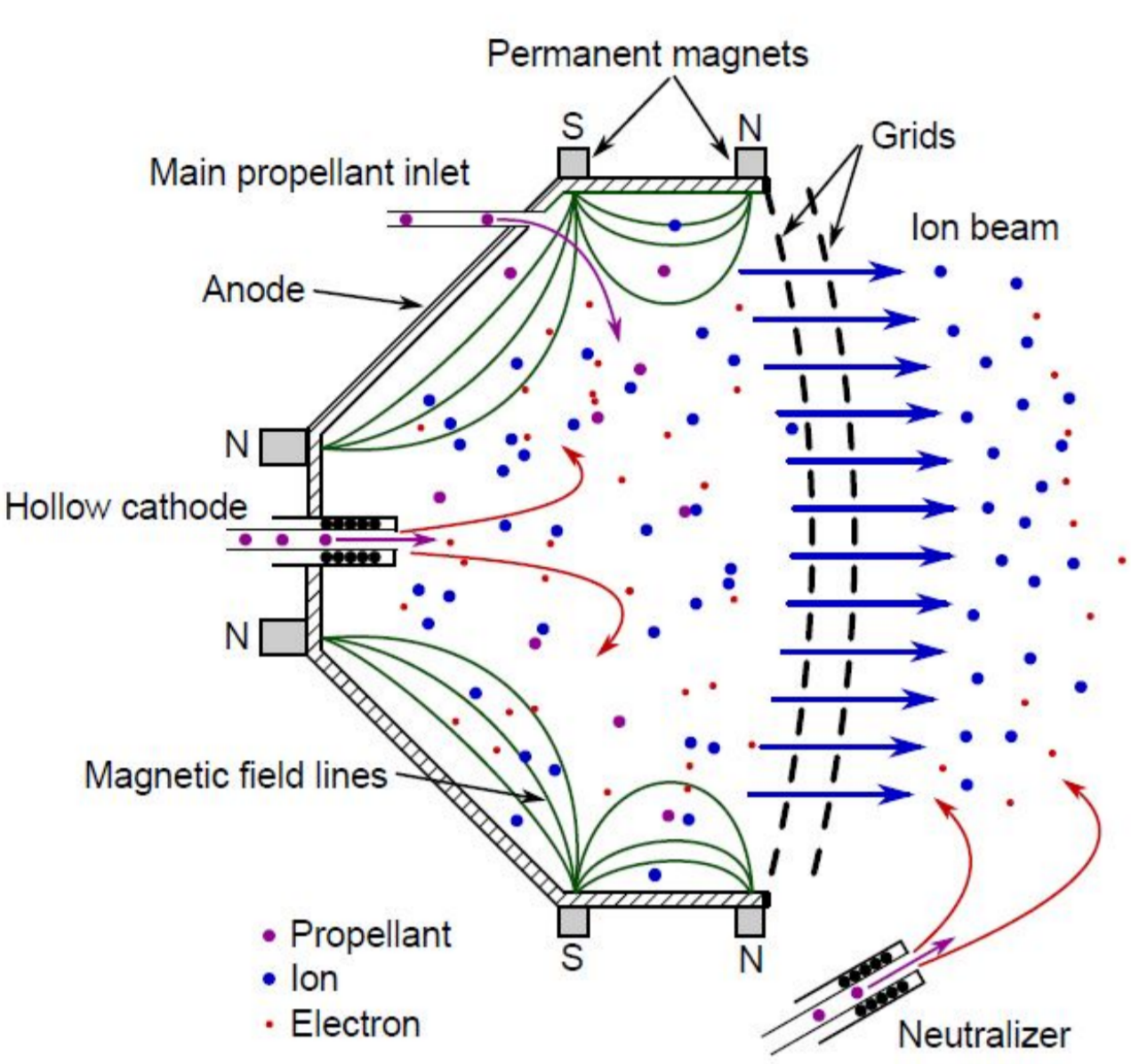
PEGASES is an ion-ion thruster concept developed at LPP for over ten years. The neutralization of the plume by alternate negative and positive ion extraction leads to a thruster design where no external neutralizer is required anymore. First 0D fluid models - or global models - of plasma thrusters were developed for DC ion thrusters operated with Xenon [1]. The model was extended to RF gridded thrusters including more complex molecular iodine chemistry [2]. Recently, neutral gas heating by ion acceleration in the sheath was added to the model, which has a very large influence on the neutral power balance.

Following the description of collisionless heating in inductively coupled RF plasma provided in [3], stochastic heating was also taken into account both through an effective collision frequency, and a heating term in the electron power balance. Refining the global model leads to a better predictability of the thruster efficiency. Both numerical PIC simulations and experiments are in progress to validate the analyses that were conducted here.

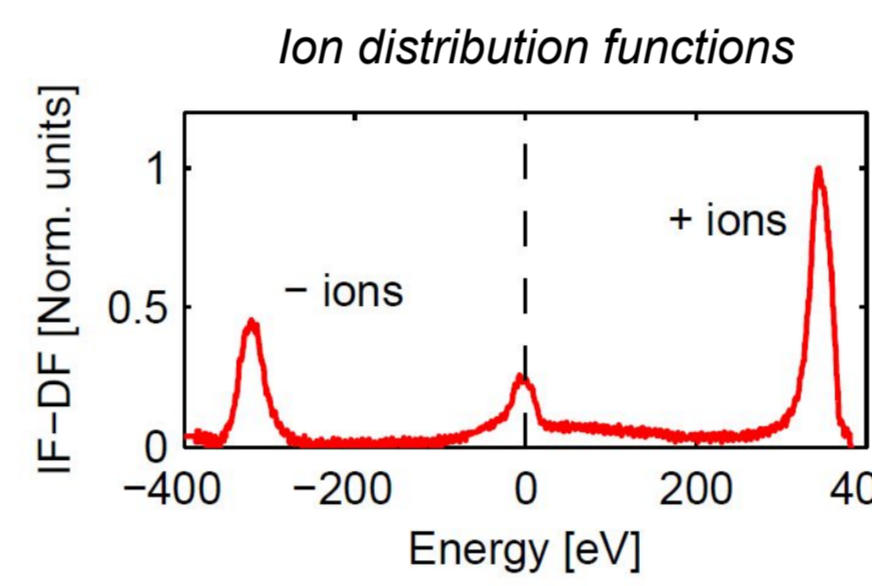
[1] P. Chabert, J. Arancibia Monreal, J. Bredin, L. Popelier, and A. Aanesland. Physics of Plasma, July 2012.
 [2] P. Grondein, T. Lafleur, P. Chabert, and A. Aanesland. Physics of Plasmas, 2016.
 [3] M. A. Lieberman and A. J. Lichtenberg. Principles of Plasma Discharges and Materials Processing. Wiley, second edition edition, 2005.

From conventional gridded thrusters...

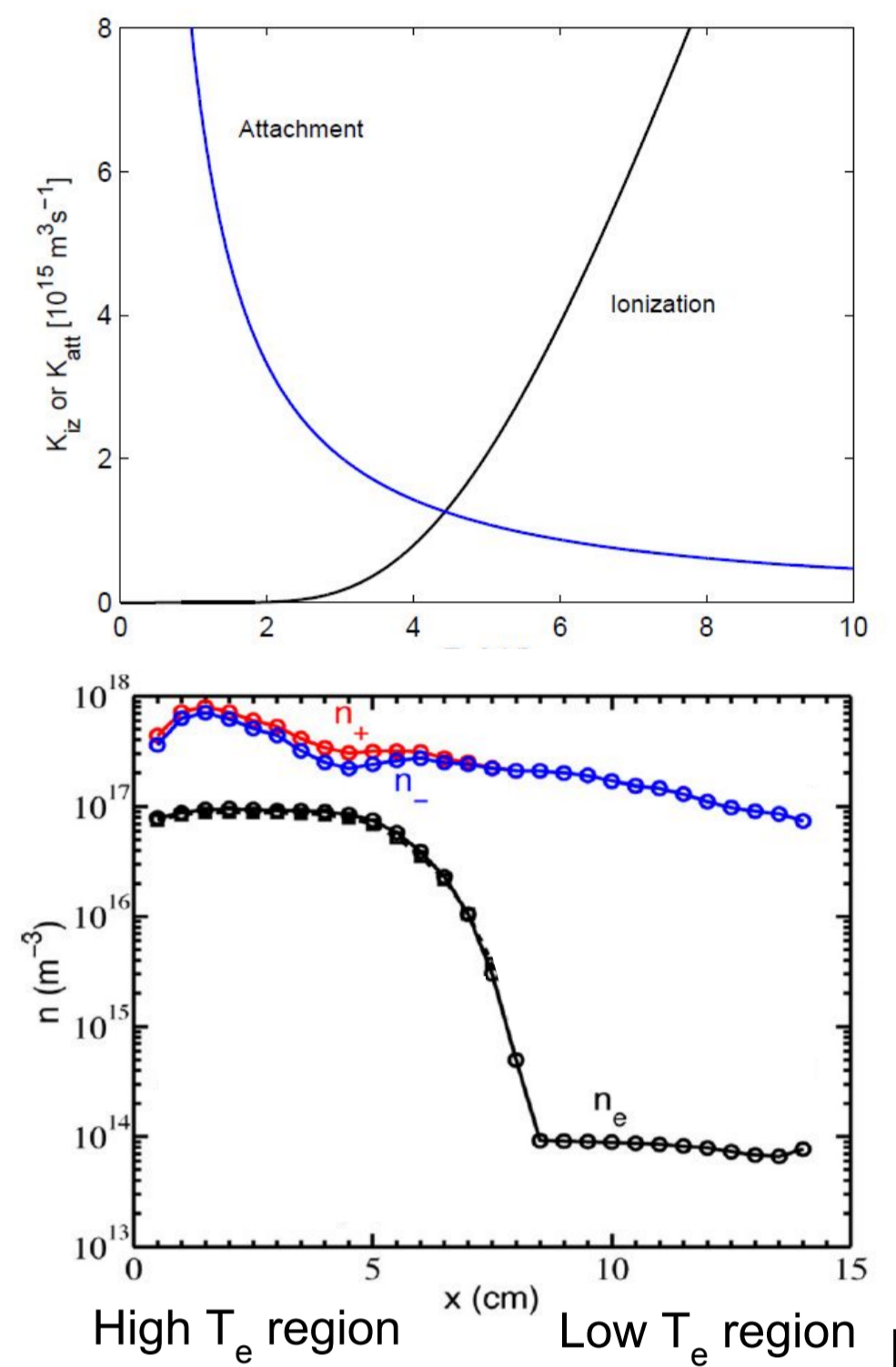
...to the PEGASES thruster



- Conventional gridded thrusters use a DC voltage on the grids, and need to reinject electrons into the exhaust plume to prevent charge buildup on the spacecraft.
- PEGASES uses electronegative gases to produce positive and negative ions.
- An RF voltage is applied to the acceleration grids, and both positive and negative ions are alternately extracted and accelerated.
- The beam packets mix downstream allowing neutralization to occur without needing an electron-emitting neutralizer.



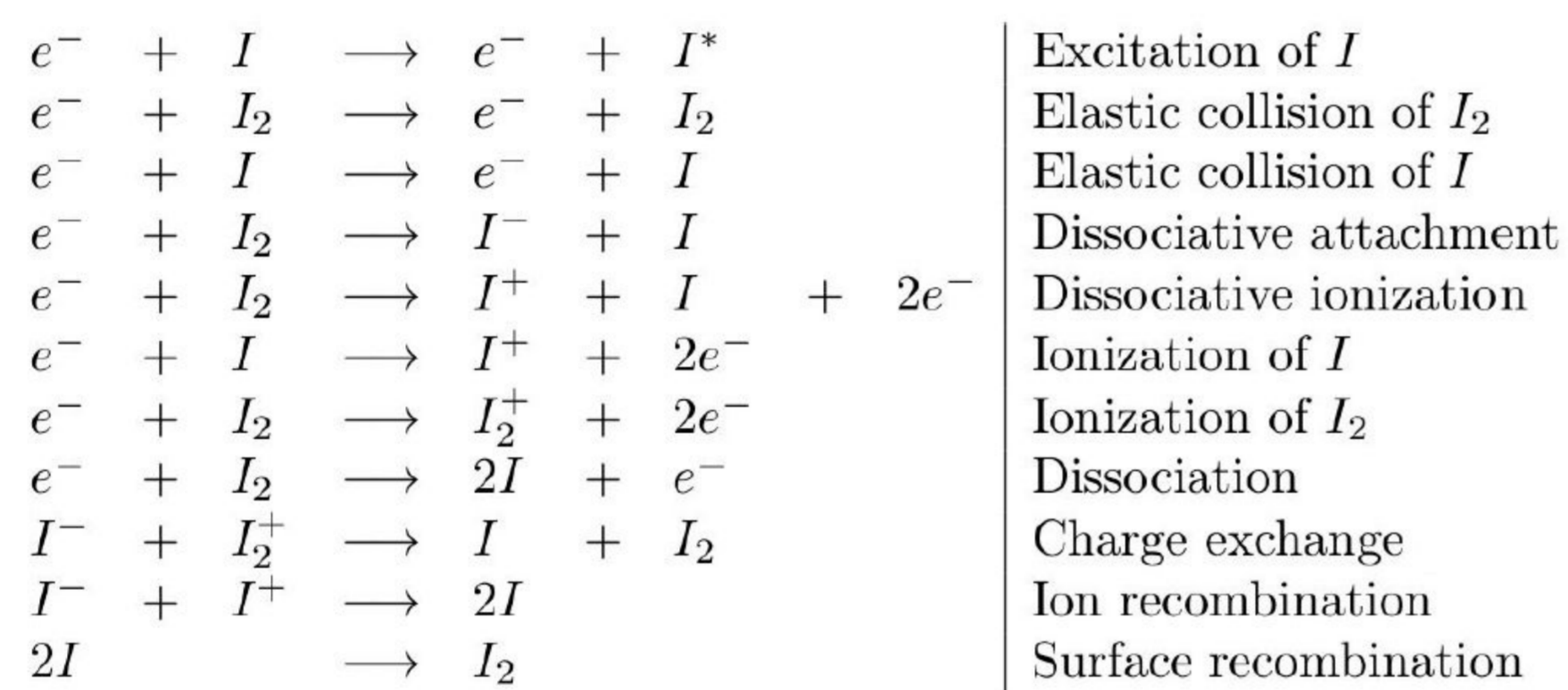
- The magnetic filter enhances electron confinement and causes a drop in electron temperature downstream.
 - Attachment increases and ionization decreases in the magnetic filter
- Electronegativity up to 1000 was reached experimentally.



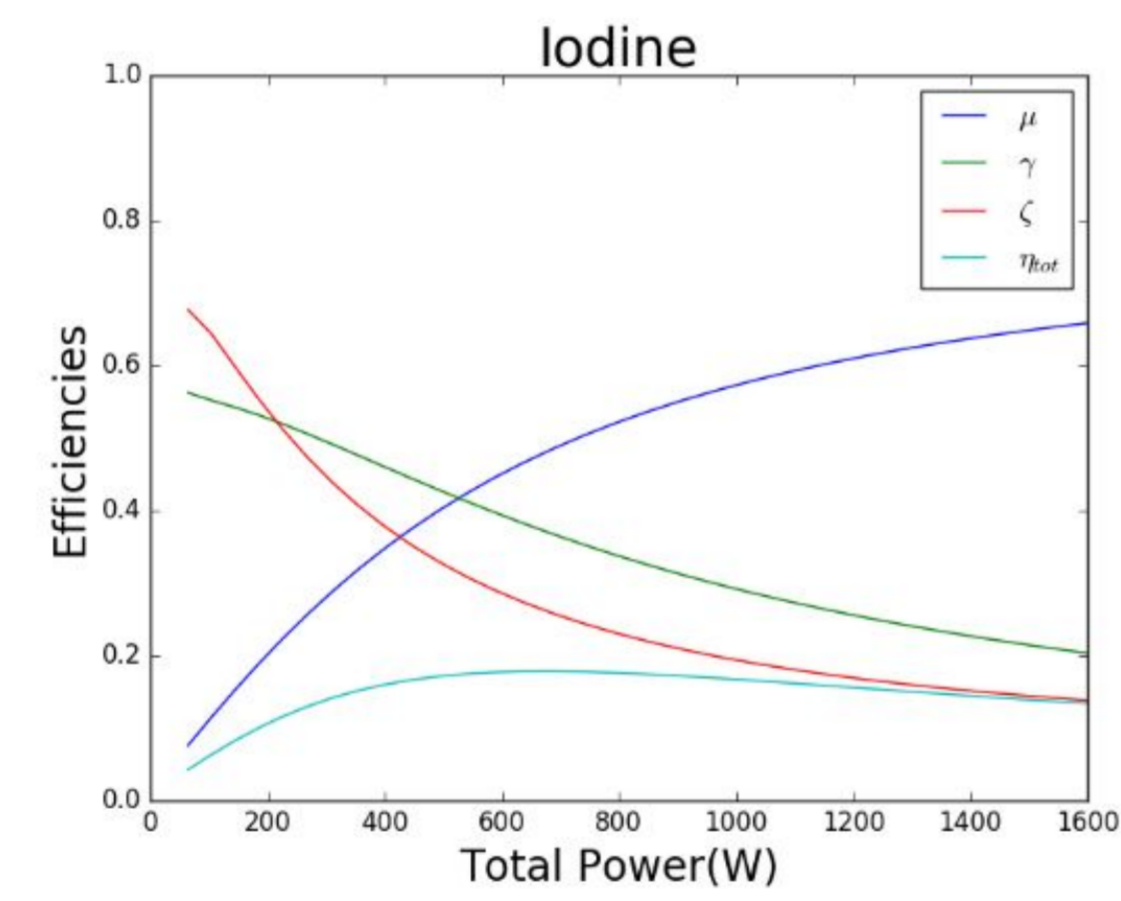
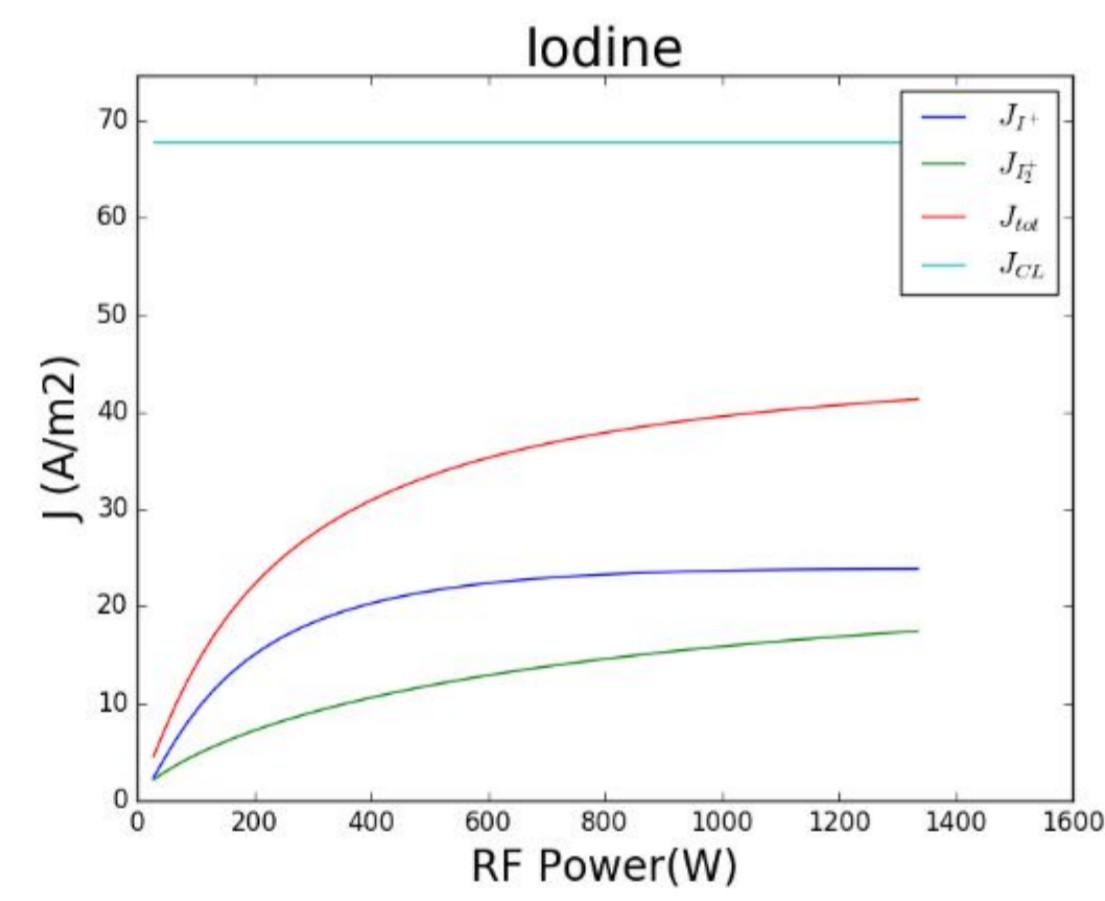
Global model

Mean parameters of the plasma and global thruster performances are estimated by fast numerical solver.

- Particle balance for each species
- Gas heating
- Electron power balance



A kinetic model was developed to simulate the iodine chemistry, including RF power injection, power loss to the walls, and grid extraction.



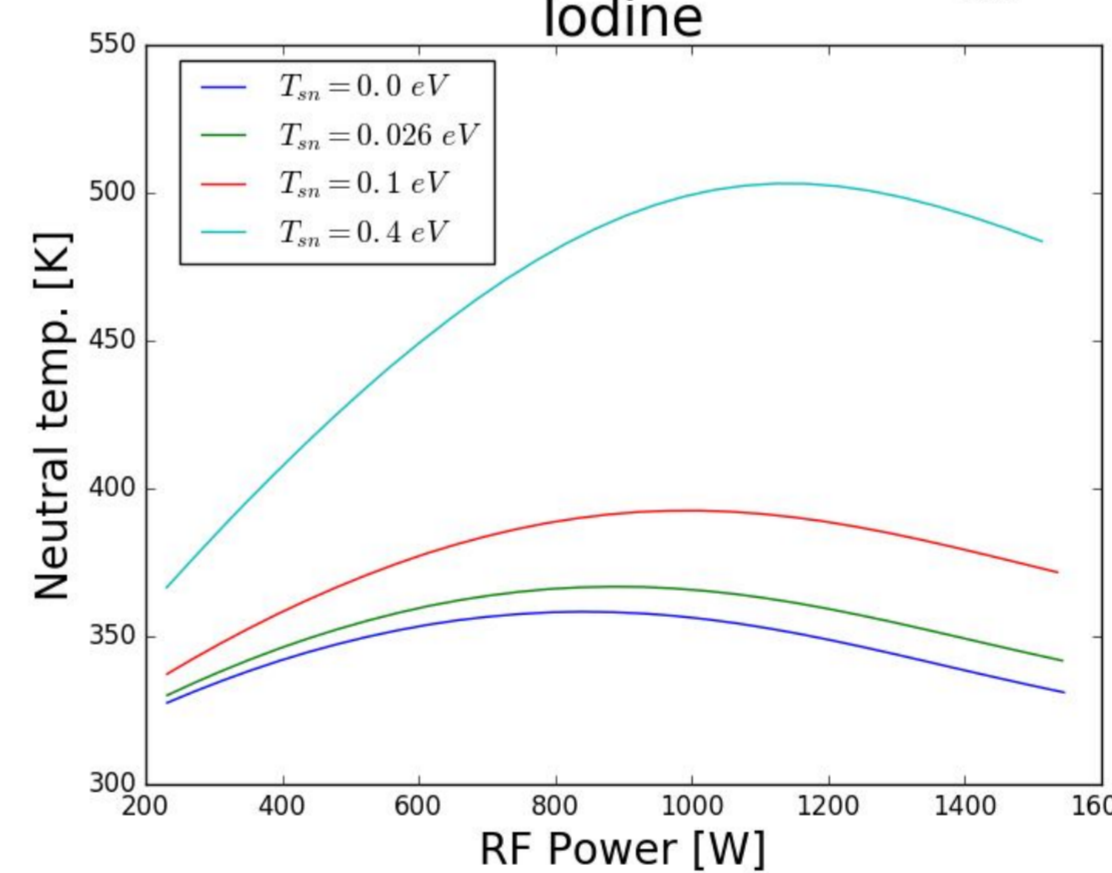
Neutral gas heating by ion acceleration in the sheath

$$\frac{d}{dt} \left(\frac{3}{2} (n_I + n_{I_2}) e T_g \right) = 3 m_e e (T_e - T_g) n_e \left(\frac{n_{I_2}}{m_{I_2}} + \frac{n_I}{m_I} \right) K_{el} + \frac{1}{4} n_e (n_{I_2} m_{I_2} u_{BI_2}^2 K_{in_{I_2}} + n_I m_I u_{BI}^2 K_{in_I}) - \kappa \left(\frac{T_g - T_{g0}}{\Lambda_0} \right) \frac{A}{V} + e T_{sn} \left(n_{I_2^+} u_{BI_2} + n_{I^+} u_{BI} \right) \frac{A_{eff} f_p}{V}$$

Power balance equation for neutrals

Effective surface for I^+ / I_2^+ production by I^+ / I_2^+ losses

Reinjected neutral temperature after ion collision to the walls



- Small fraction of the ion energy gained in the sheath contributes to gas heating
- Both constant reinjection temperature approach and accommodation coefficient approach are investigated
- Experiments are planned to determine T_{sn} in an iodine plasma with more accuracy using Two-Photon Absorption Laser-Induced Fluorescence (TALIF) methods.

Stochastic heating in an inductively-coupled plasma

- Transverse electric field gradient in the skin depth of an inductive discharge
- RF perturbation of the electron velocity distribution function at small space scales induces collisionless heating
- Induced surface power flux depends on the skin depth δ :

$$\bar{S}_{stoc} = \frac{m n_e}{\bar{v}_e} \left(\frac{e \tilde{E}_0 \delta}{m} \right)^2 \mathcal{I}(\alpha)$$

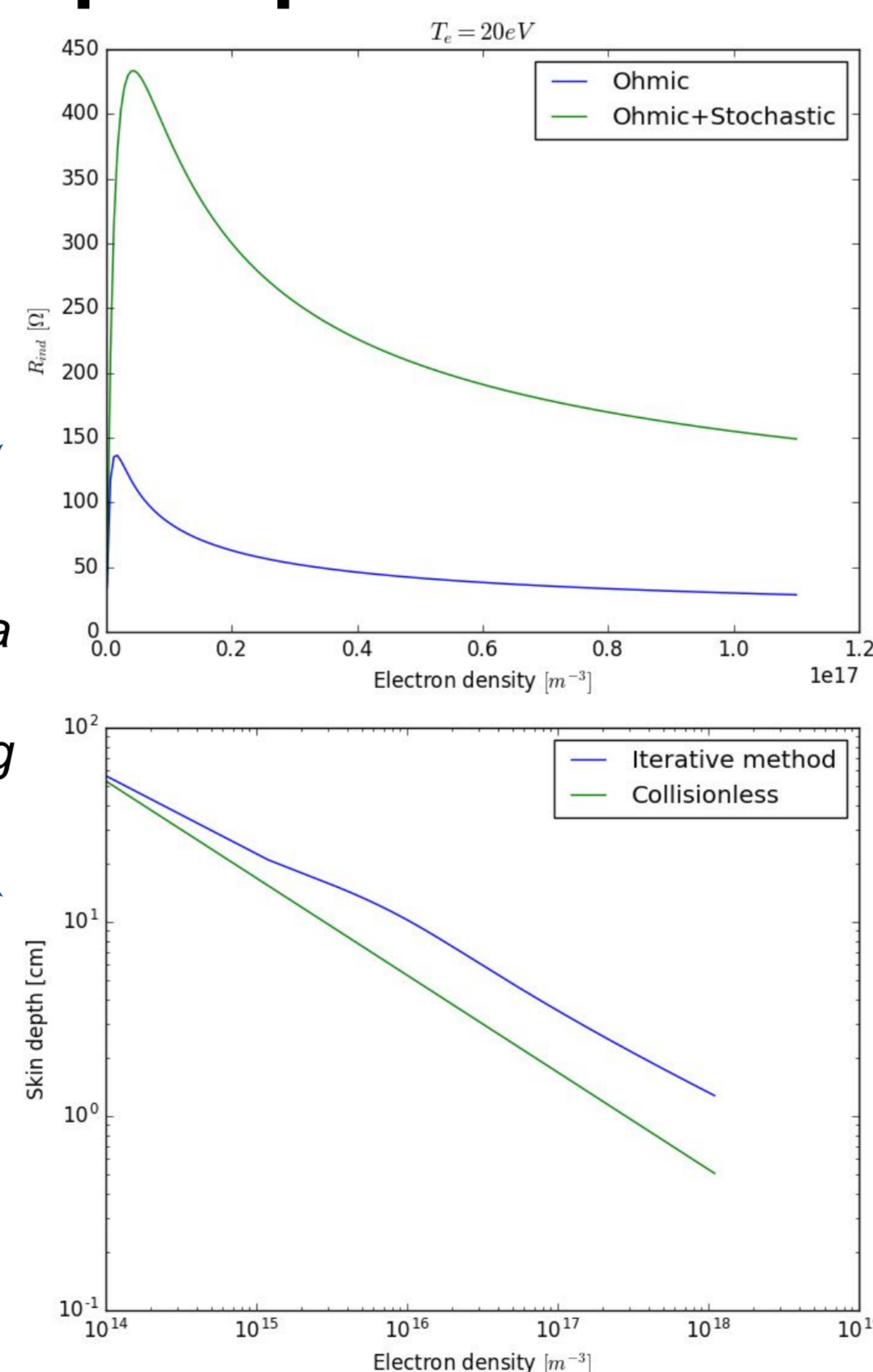
- Equivalent collision frequency

$$\nu_{stoc} \approx \frac{C_e(\alpha) \bar{v}_e}{4 \delta}$$

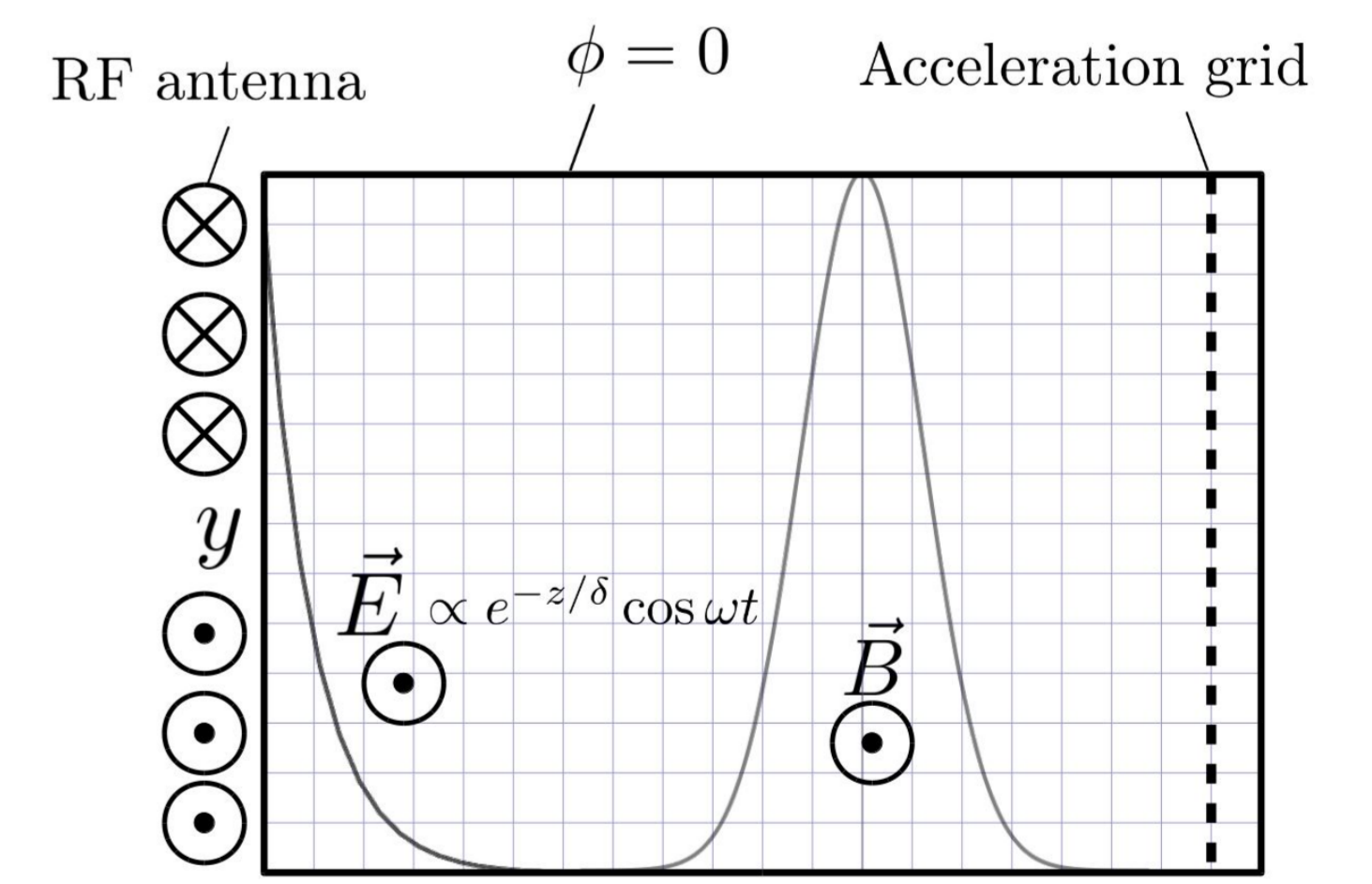
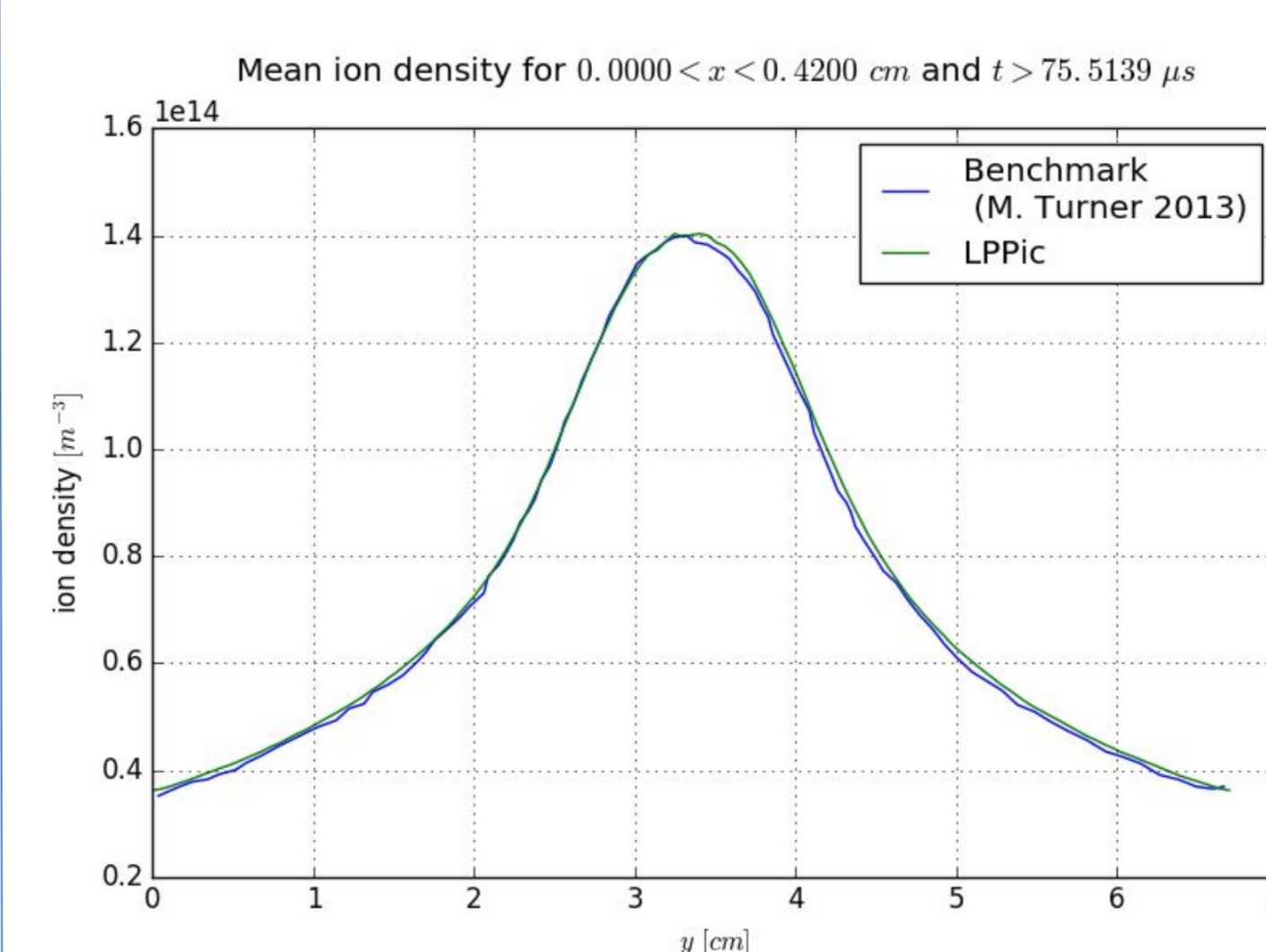
- The skin depth accounting for collisionless heating was calculated iteratively
- Plasma resistance rises by a factor 3 due to stochastic heating

Neutral gas density	n_g	1.0×10^{19}	m^{-3}
RF frequency	ω	13.56	MHz

Estimated plasma response to stochastic heating



2D PIC simulations

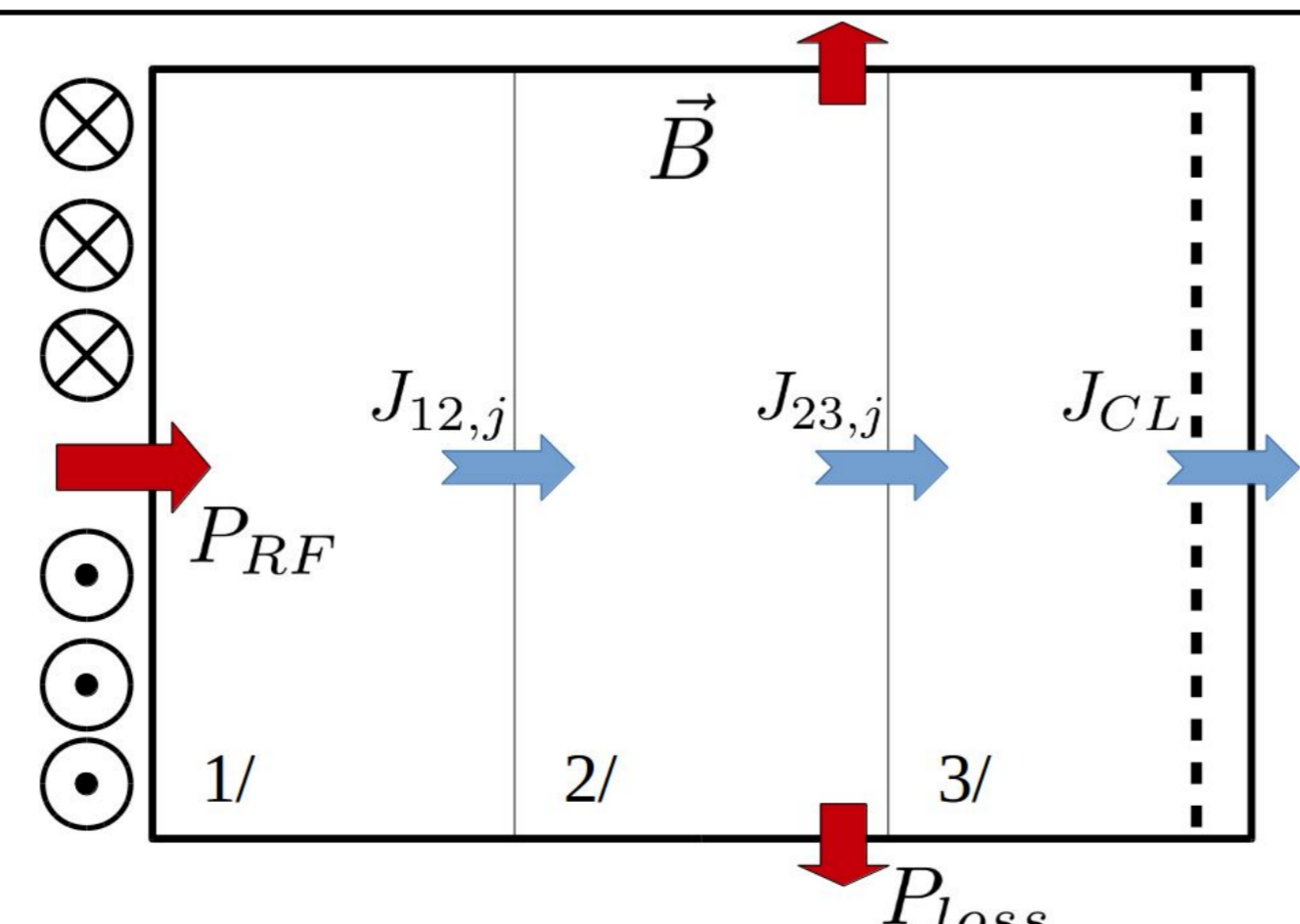


- 2D PIC code modified to run with any noble gas type propellant (Xe, He, Ar, Kr, I electropositive)
 - 1D capacitively coupled discharge benchmark validated
- After validation by the global model, simplified iodine chemistry will be implemented in the code.
- Magnetic barrier of PEGASES can be simulated with argon and iodine, for which many experiments were conducted.
- Complex geometry of acceleration grid will be implemented in the future

Future magnetised global model

Improvement of the global model to take into account electron confinement by the magnetic field.

- 3 regions:
 - RF heating
 - Electron attachment
 - Downstream extraction and acceleration
- Thermal fluxes connect the 3 regions together
- A longer effective distance for electrons in region 2/ is implemented to account for an increased confinement time.



PIC/GM connections

